

LOW SPEED vs HIGH SPEED TESTING OF LP TURBINE BLADE WAKE INTERACTION

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Introduction

- Experiments simulating blade wake interaction at high Mach numbers are needed

BUT

- Only IF they are worth it

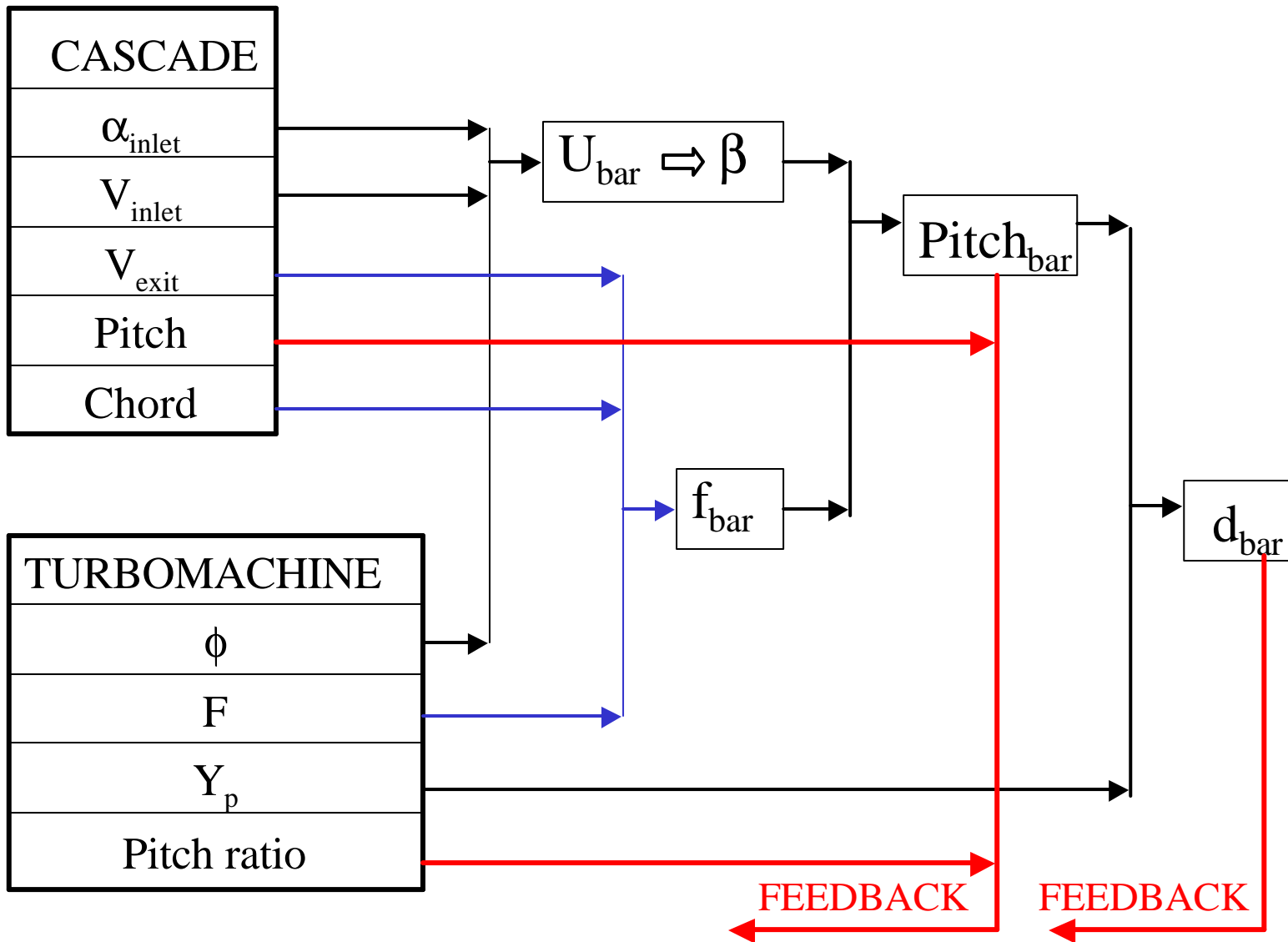
This seminar presents

- Techniques for testing of low and high speed LPT cascades
- The extent to which results from low speed are meaningful

High speed vs. low speed

	High speed	Low speed
Reynolds number	✓	✓
Mach number	✓	✗
Exit angle	✓	✓
Stream tube divergence	✗	✗
Chord	50mm	200mm
Bar passing frequency	3KHz	60Hz
Bar diameter	0.4mm	2mm

Choosing the size

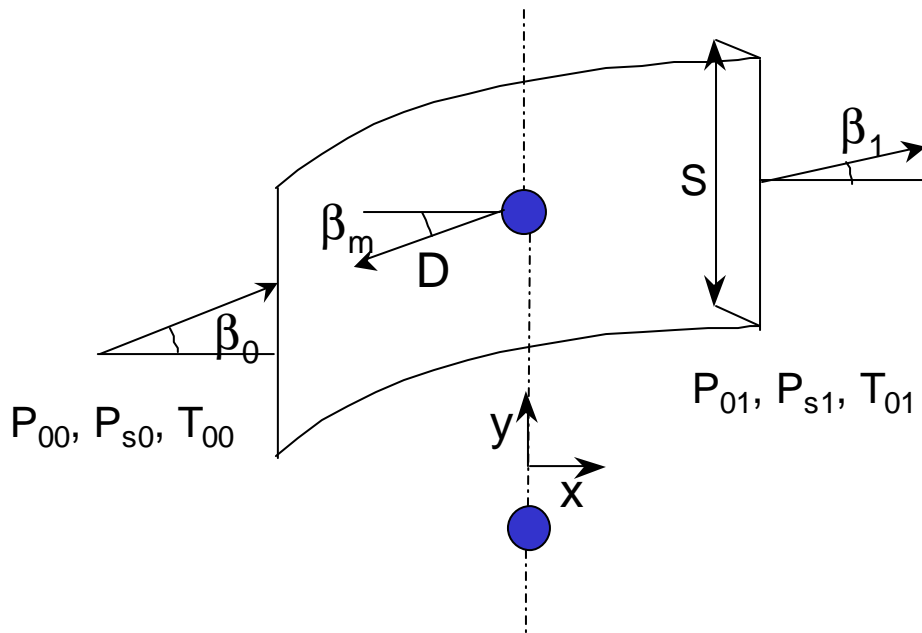


Determination of bar losses

Stagnation pressure across the bar row changes due to

- Losses due to the drag of the bars
- Work input

Mixed out conditions downstream the bars are needed

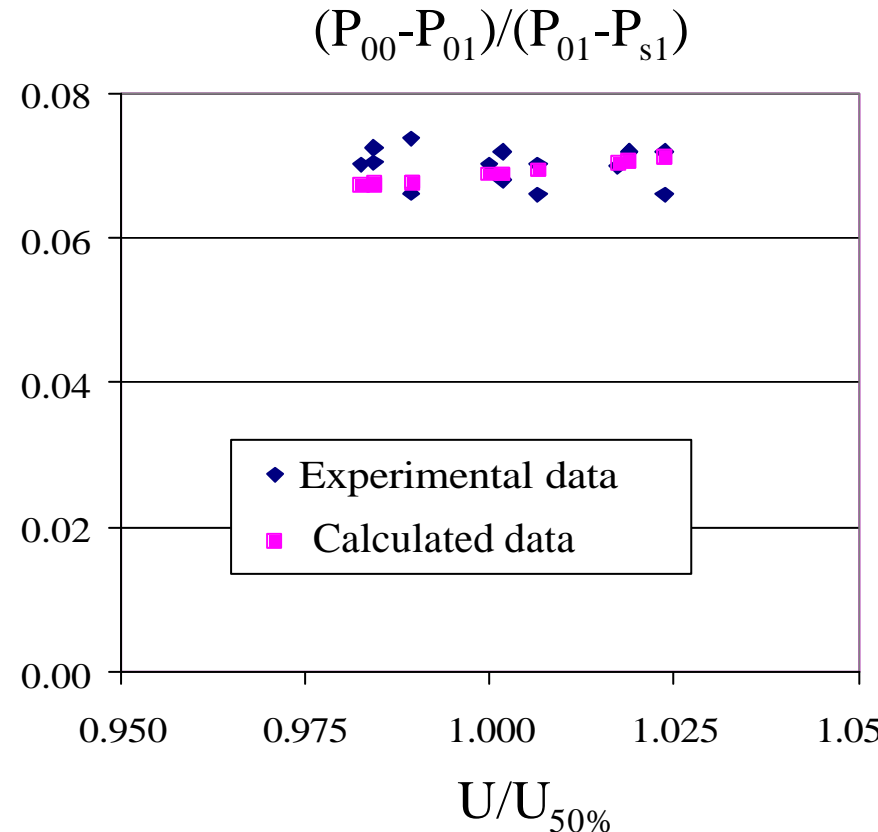
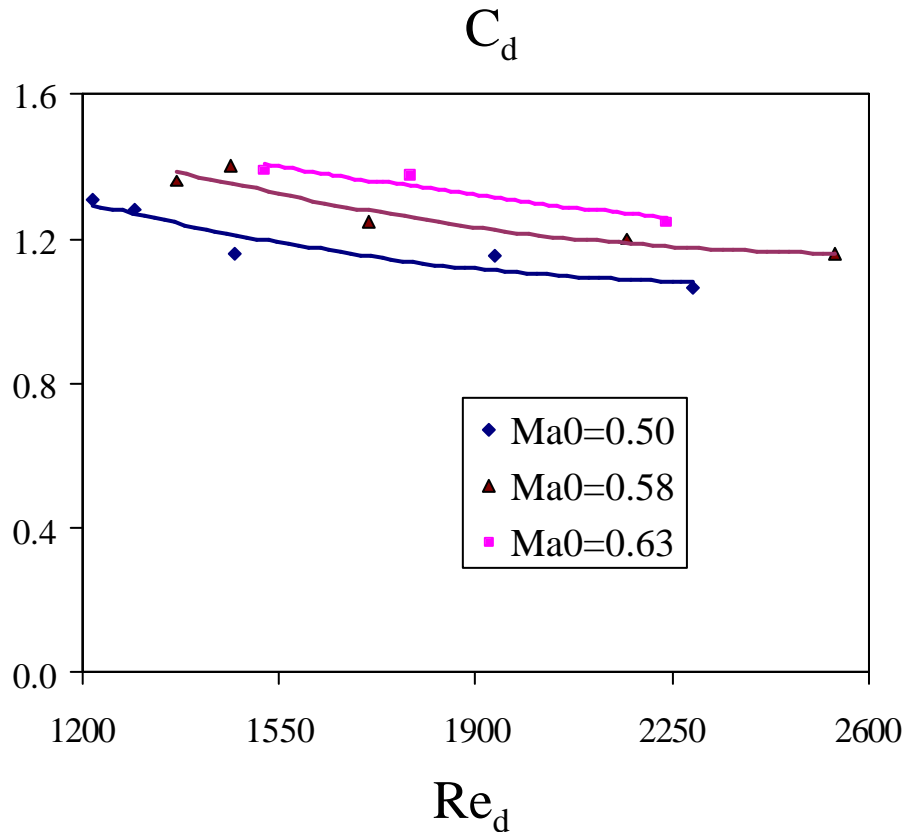


Calculation procedure

Frame of reference
fixed to the bars

$$D = d \cdot C_d \cdot \begin{cases} 1/2 \cdot \rho \cdot W^2 & \text{Low speed} \\ (P_{00} - P_{s0}) & \text{High speed} \end{cases}$$

Determination of bar losses

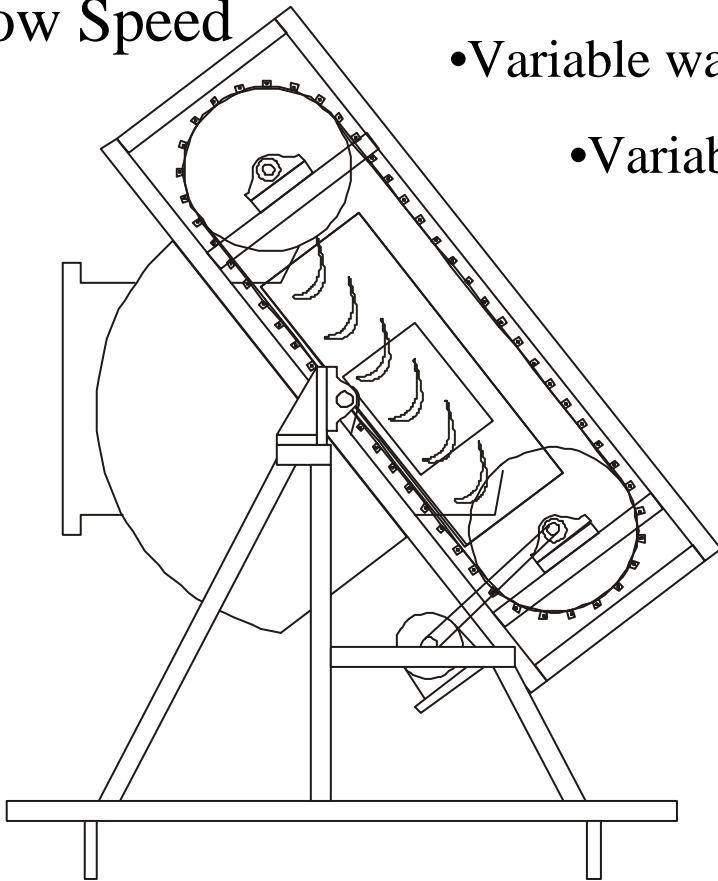


- A good estimation of C_d is needed
- The calculation procedure is validated

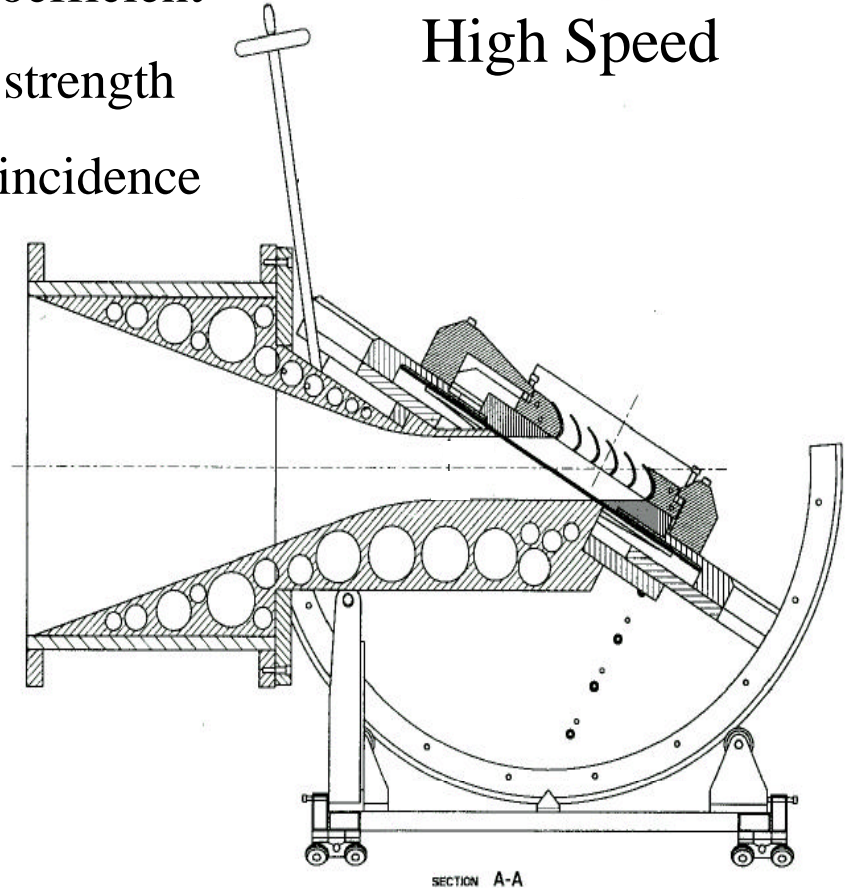
Facilities

- Variable wake passing frequency
- Variable inlet flow coefficient
- Variable wake strength
- Variable incidence

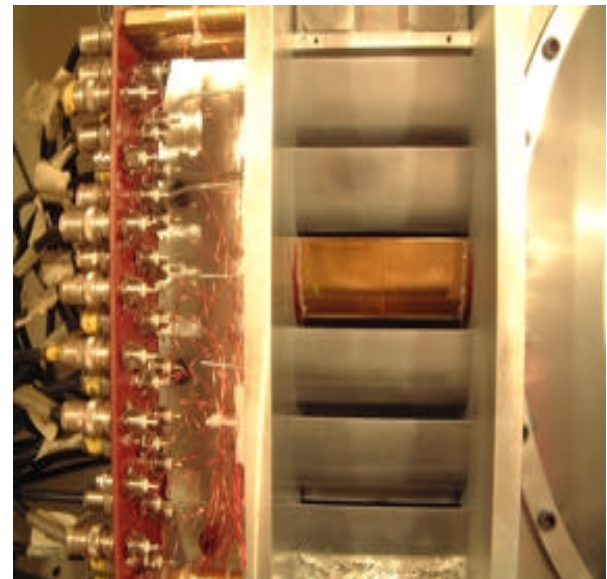
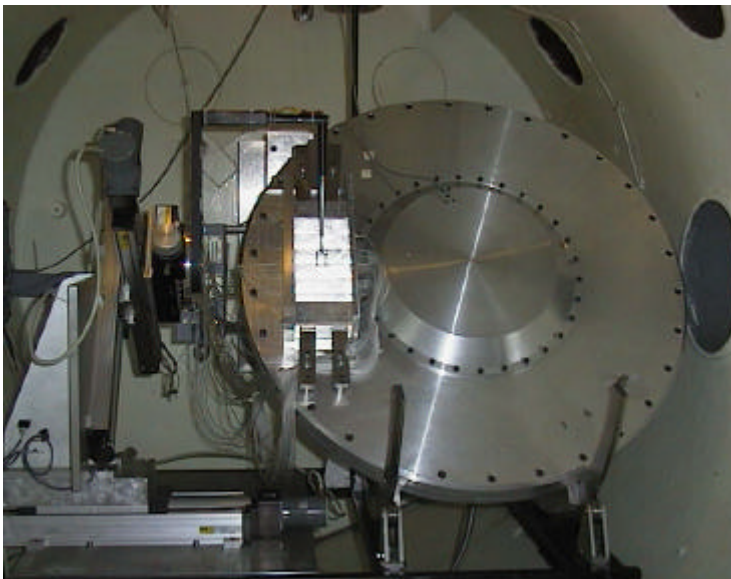
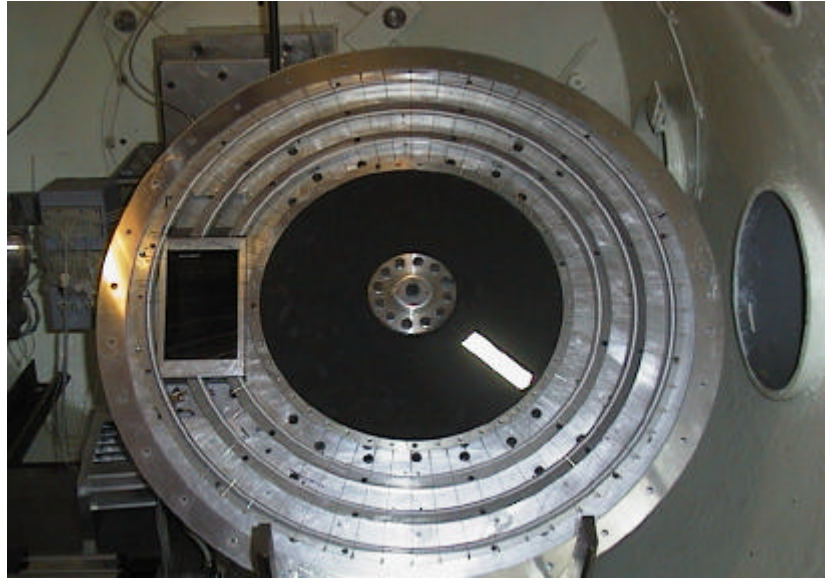
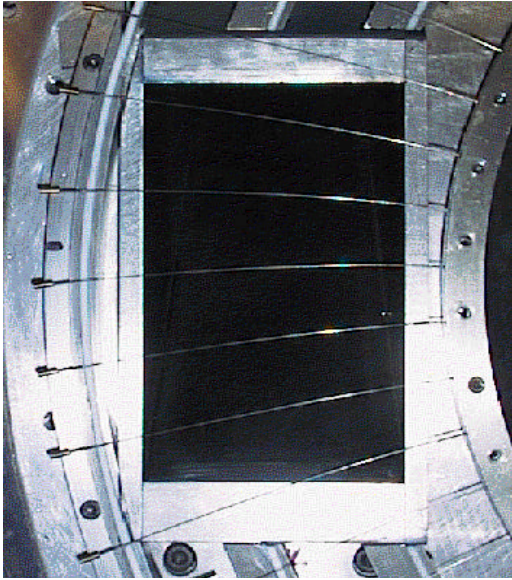
Low Speed



High Speed



Facilities. High speed rotating bar rig

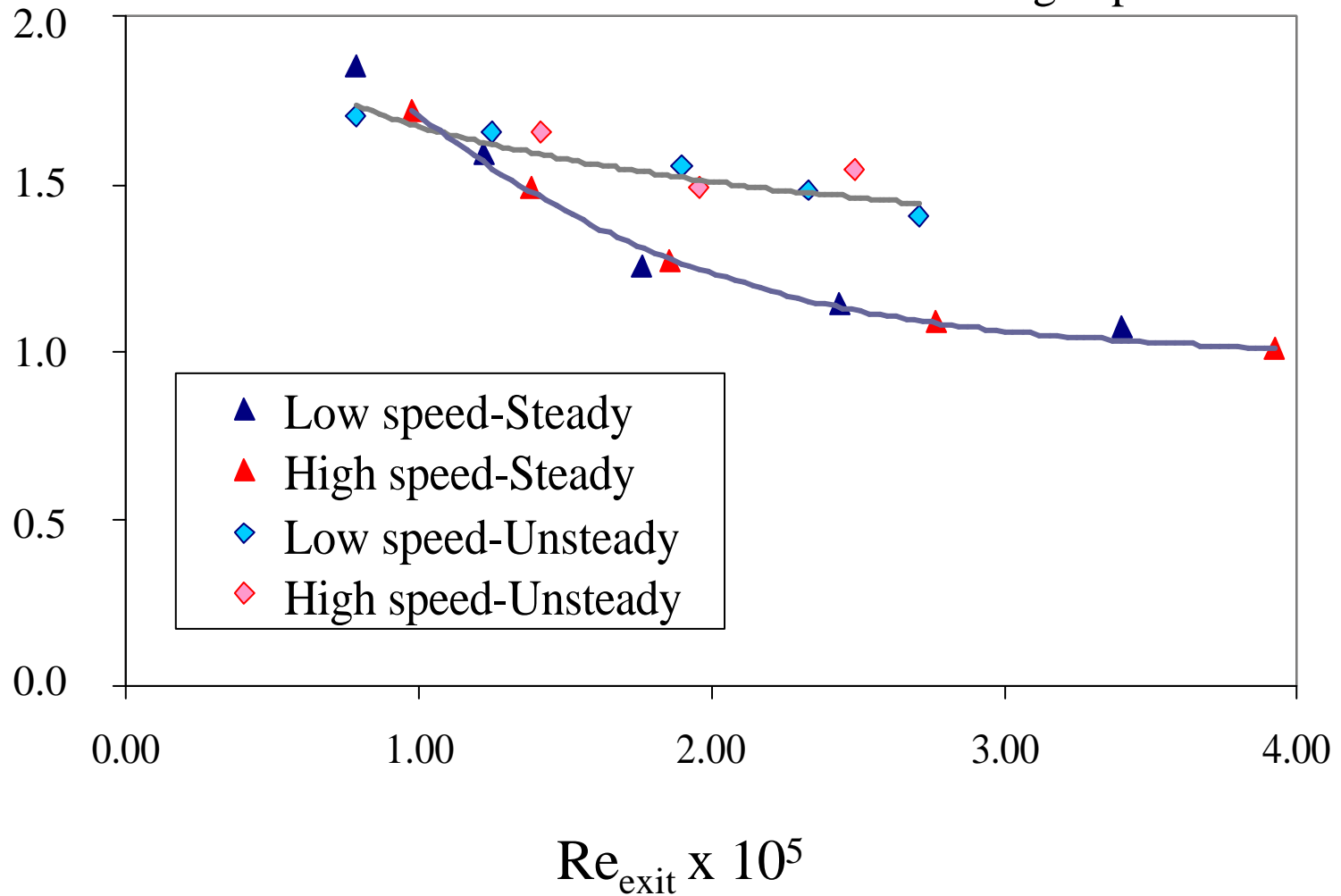


Results

Low speed: Y_p

Losses

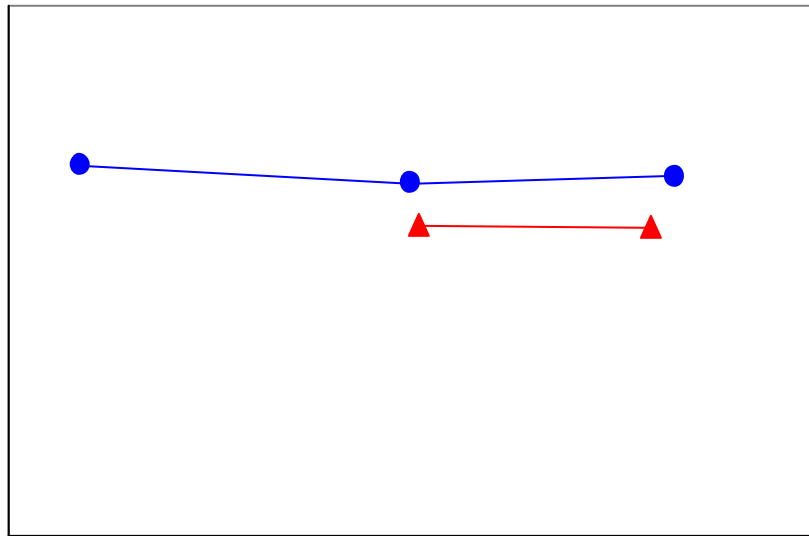
High speed: KSI



Results

High lift, high turning profile

KSI/KSI_{design}



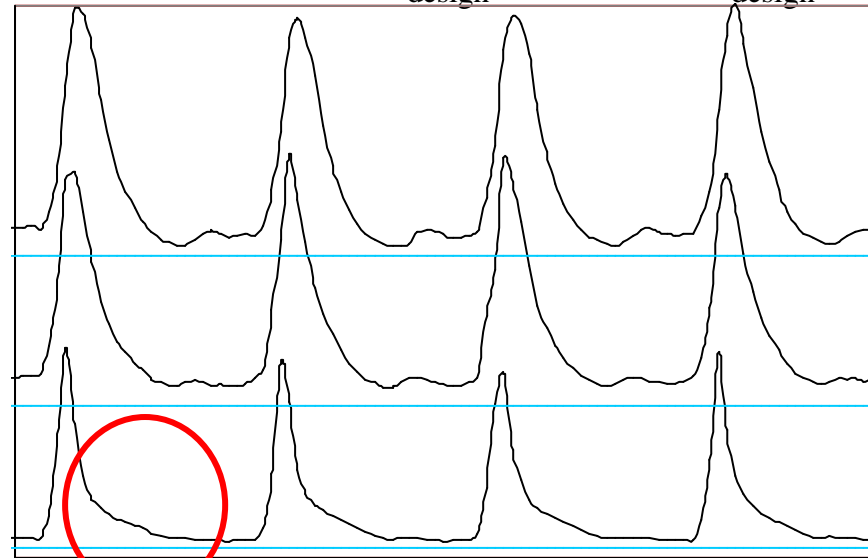
0.43

0.62

Ma/Ma_{design}

0.80

Non-dimensional ensemble averaged quasi shear stress $Ma=Ma_{design}$, $Re=0.65Re_{design}$



Bar passing period

- Independency of Mach number below the transonic range
- Existence of becalmed region following the wake induced path

Conclusions

High speed testing presents greater challenges.

At all speeds below transonic:

- Same quantitative results are obtained for loss
- Same wake interaction phenomena are found

Low speed approach is valid below the transonic range
and it is EASIER

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