

The 16th Symposium on Measuring Techniques in Transonic and Supersonic Flow
in Cascades and Turbomachines

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REYNOLDS STRESS MEASUREMENT WITH 1D LDA

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Outline

- Motivation for technique
 - Capital cost
 - Equipment availability
 - Processing technique to extend measurements possible for 1D LDA
- Derive Equations
 - Mean flow and Turbulent statistics
 - Improve mean and turbulent data using least squares
- Demonstrate Technique
 - Back to back comparison of rotated 1D and 2D LDA
 - Show effect of number of angles and number of samples on measurement quality

Reynolds Stresses

- Reynolds stress (2D)

$$\overline{uv} = \frac{1}{N} \sum_{i=1}^N \frac{(U - \bar{U})(V - \bar{V})}{UV}$$

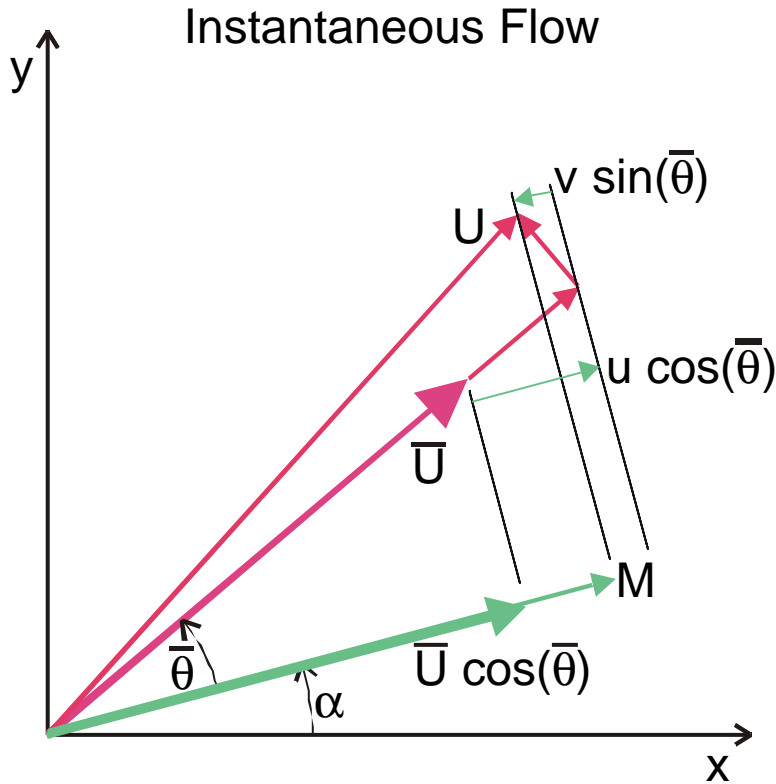
- Require two simultaneous components of velocity
- Calculate covariance
- For steady or phase locked flows we can make the measurements independently and recombine them to get the covariance

Basis of Technique

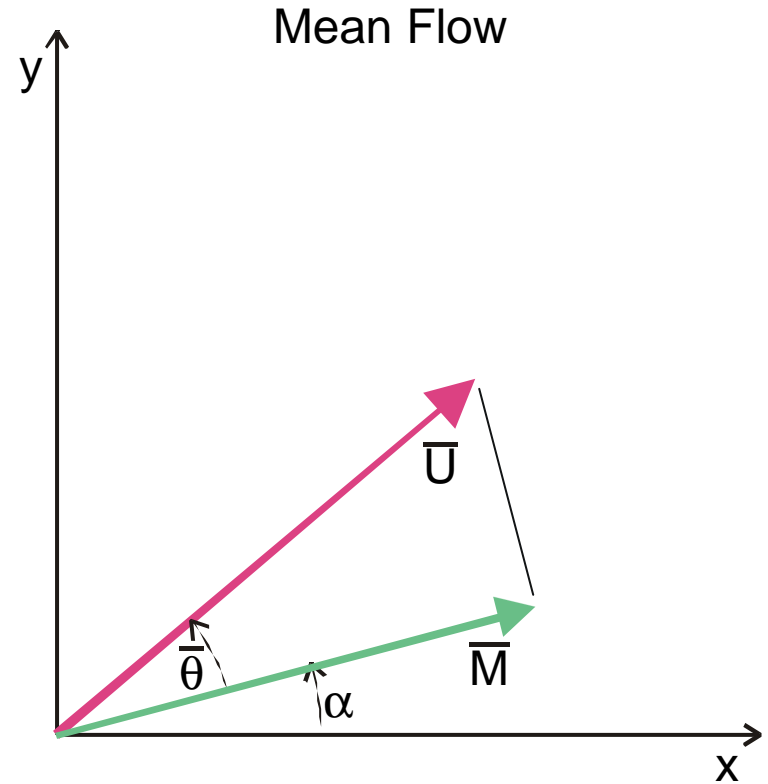
- LDA measures velocity component in the plane of the intersecting beams
- Pure cosine relationship between measured velocity and probe angle relative to flow

$$M = U \cos(\mathbf{q})$$

Derivation



$$M = (\bar{U} + u) \cos(\bar{\theta}) - v \sin(\bar{\theta})$$



$$\bar{M} = \bar{U} \cos(\bar{\theta})$$

subtract mean from each sample, then square, then average

$$\overline{m^2} = \overline{u^2} \cos^2 \bar{q} + \overline{v^2} \sin^2 \bar{q} - 2\overline{uv} \sin \bar{q} \cos \bar{q}$$

If $\overline{m^2}$ and \bar{q} known then 3 unknowns

thus 3 probe orientations required

Find the mean Flow

- Fit measured data onto known angular response of LDA
- Minimise error S if more than 2 probe angles used
 - Non-linear in variables \bar{U} and $\bar{\mathbf{q}}$
 - Use numerical approach
 - eg Levenberg-Marquardt algorithm

$$S = \sum_{i=1}^N (\bar{M}_i - \bar{U} \cos(\bar{\mathbf{q}}_i))^2$$

Find turbulent flow

- Functional relationship is

$$\overline{m^2} = \overline{u^2} \cos^2 \overline{\mathbf{q}} + \overline{v^2} \sin^2 \overline{\mathbf{q}} - 2\overline{uv} \sin \overline{\mathbf{q}} \cos \overline{\mathbf{q}}$$

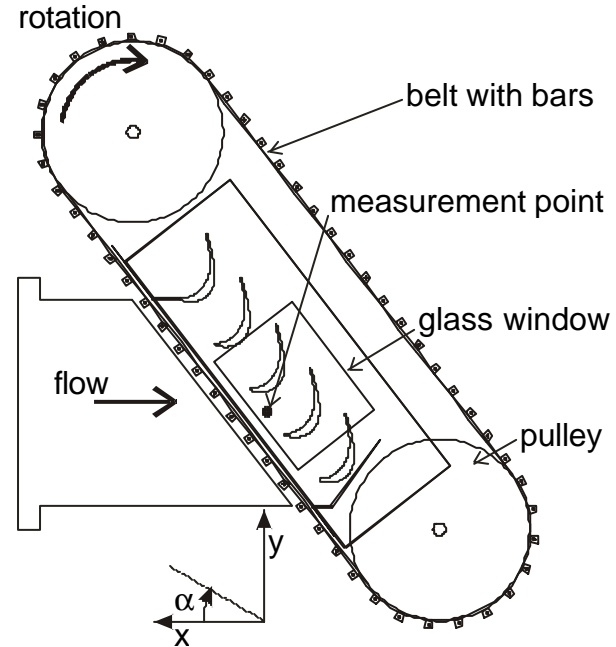
- know $\overline{\mathbf{q}}$ (previous slide) and $\overline{m^2}$ (measurement) from more than 3 probe orientations
- Use least squares fit (linear) to improve data

$$\sum_{i=1}^N A(\overline{\mathbf{q}}_i) \cdot \begin{bmatrix} \overline{u^2} \\ \overline{v^2} \\ \overline{uv} \end{bmatrix} = \sum_{i=1}^N \overline{m_i^2} \begin{bmatrix} \cos^2(\overline{\mathbf{q}}_i) \\ \sin^2(\overline{\mathbf{q}}_i) \\ \cos(\overline{\mathbf{q}}_i) \sin(\overline{\mathbf{q}}_i) \end{bmatrix}$$

$$A(\overline{\mathbf{q}}_i) = \begin{bmatrix} \cos^4(\overline{\mathbf{q}}_i) & \cos^2(\overline{\mathbf{q}}_i) \sin^2(\overline{\mathbf{q}}_i) & \cos^3(\overline{\mathbf{q}}_i) \sin(\overline{\mathbf{q}}_i) \\ \cos^2(\overline{\mathbf{q}}_i) \sin^2(\overline{\mathbf{q}}_i) & \sin^4(\overline{\mathbf{q}}_i) & \cos(\overline{\mathbf{q}}_i) \sin^3(\overline{\mathbf{q}}_i) \\ \cos^3(\overline{\mathbf{q}}_i) \sin(\overline{\mathbf{q}}_i) & \cos(\overline{\mathbf{q}}_i) \sin^3(\overline{\mathbf{q}}_i) & \cos^2(\overline{\mathbf{q}}_i) \sin^2(\overline{\mathbf{q}}_i) \end{bmatrix}$$

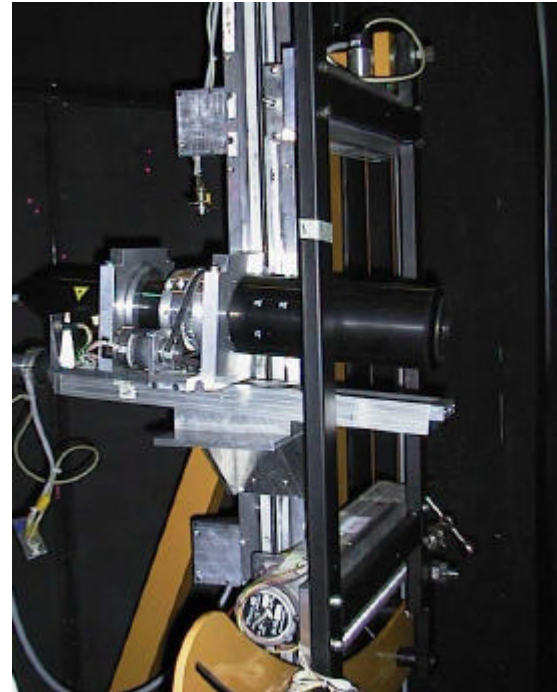
Application of Technique

- Measured wake of passing bar
 - simple flow
 - phase locked by trigger
- 120 000 samples at each orientation
- 128 phase bins for ensemble averaging
- 2D measurement at multiple probe orientations, extract 1D data and compare with 2D data



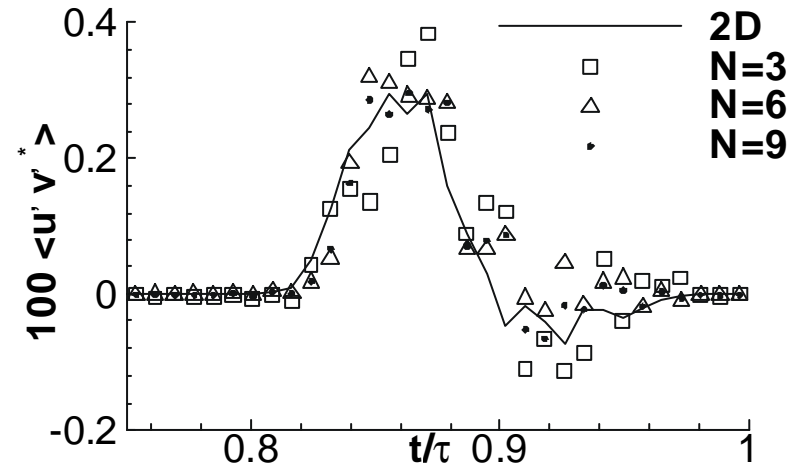
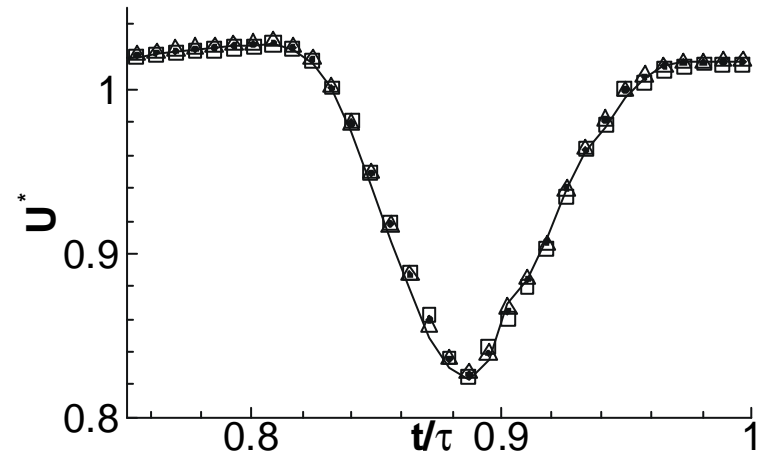
LDA Equipment

- 5W Argon-Ion laser
- 1D used 514nm, 2D also 488nm beam
- Dantec BSA signal processor
- 2D 85mm Probe
- 500 mm focal length
- Stepper motor traverse for probe rotation



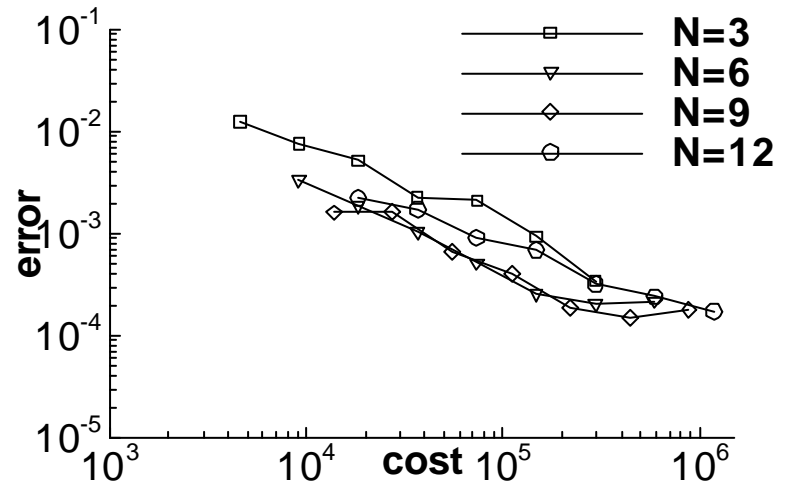
Evaluation of rotated 1D LDA

- mean
 - good agreement for all numbers of angles (N)
- turbulent statistics
 - improve with number of samples (number of angles)



Error vs. Cost

- Estimate error in $\langle uv \rangle$ as variance of 1D from 2D results
- Cost is total number of samples used
- high error at all costs for $N=3$
 - shows benefit of least squares approach
- highest N not necessarily lowest error
 - need to have good estimates for mean and variance of each measurement angle



Conclusion / Discussion

- Exact angular response
 - No need for angular calibration nor small angle assumptions of equivalent HWA technique
- Spatial resolution of LDA maintained
- No directional ambiguity (LDA)
- Reduced capital cost traded against increased acquisition time

Other Application

- dissipation measurement

$$C_D = \frac{1}{rU_\infty^3} \int_0^d \frac{1}{T} \mathbf{t}_{yx} \frac{du}{dy} dy$$

$$\mathbf{t}_{yx} = m \frac{du}{dy} - r \overline{u'v'}$$

