

# 1 Predicting Failure in Glass—A General Crack Growth Model

2 M. Overend<sup>1</sup>; G. A. R. Parke<sup>2</sup>; and D. Buhagiar<sup>3</sup>

3  
4 **Abstract:** The successful design of glass as a structural element depends mainly on the ability to predict failure with accuracy and ease.  
5 Over the last 30 years various failure prediction models have been put forward for determining the load bearing capacity of glass, some  
6 of which have been adopted in national codes of practice. The differences between these models translate into a wide range of glass  
7 strength and glass thickness values in glass design. This paper compares the mathematical formulations of a number of existing failure  
8 prediction models, and the differences between these models are identified and discussed. From these comparisons a general crack growth  
9 model (GCGM) based on established statistical failure theory with linear elastic fracture mechanics is proposed. The performance of the  
10 existing models and the proposed GCGM is compared by physical and numerical investigations. The proposed model is shown to provide  
11 a basis for an accurate and automated method for determining the tensile strength of glass subjected to static loads and valid for any  
12 geometry and support condition.

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15  
16

## 17 Introduction

18 Glass structures have evolved from traditional curtain wall glaz-  
19 ing, in which the glass is supported along two or four edges by a  
20 metal framework, to the current point-supported structural glass  
21 assemblies, where the glass plates are connected to each other and  
22 to a supporting structure by discrete clamped or bolted fixings  
23 usually located toward the corners of the glass panels. Point-  
24 supported glass facades are often top hung, i.e., the upper plate of  
25 glass carries the self-weight of the plate below; therefore the glass  
26 plates in top-hung structural glass facades are subjected to a com-  
27 bination of lateral wind loads and in-plane load. Furthermore,  
28 various glass elements are increasingly being used as load bearing  
29 elements in locations other than the façade, e.g., glass beams or  
30 fins, glass stairs, glass floors, glass balustrades, etc.

31 The lack of an accurate and user-friendly methodology for  
32 determining the strength of glass, particularly one that caters to  
33 the wide range of possible loading and support conditions, in-  
34 duces engineers to adopt large safety factors and expensive pro-  
35 totype testing. In addition, there is a lack of published research  
36 and test data on the performance of the more recent forms of  
37 structural glass assemblies.

38 Despite these shortcomings, a number of failure prediction

models have been proposed over the last 30 years. These models  
were originally developed for laterally loaded rectangular glass  
plates; however they provide a valuable source of information on  
the factors that affect glass strength and the mechanics of glass  
failure. These failure models include the pioneering work of  
Brown (Brown 1974) and the seminal work of Beason and  
Morgan (Beason 1980; Beason and Morgan 1984) that form the  
basis of the United States and Canadian codes of practice (ASTM  
1997; CAB/CGSB 1989). Two of the more recent models are: the  
model proposed by Sedlacek et al. (1995) that forms the basis of  
the European code of practice (CEN 1997); and the crack growth  
model put forward by Fischer-Crippps and Collins (1995). The  
common approach of all these models is that maximum stress  
oriented theories are unable to portray the strength of glass ac-  
curately and that an accurate determination of the strength of glass  
should be achieved by relating the probability of failure to the  
factors affecting Griffith flaw characteristics. This is widely  
accepted as the most accurate approach by the glass design com-  
munity, however the complexity of these models makes them un-  
attractive for manual computation.

This paper summarizes the basic mechanics of glass failure  
with respect to the various factors that affect glass strength and  
discusses how the above-mentioned failure models allow for  
these factors. A general failure prediction model based on estab-  
lished statistical failure theory and linear elastic fracture mechan-  
ics is also put forward in this paper. The proposed model, referred  
to as the general crack growth model (GCGM), extends the  
Fischer-Crippps and Collins (1995) model to account for variations  
in the maximum and minimum principle stresses on the surface of  
the glass and covers the use of heat strengthened and fully  
tempered glass. An automated approach is subsequently used to  
compare the predictions obtained from the existing and proposed  
failure models. Thee predictions are verified by experimental in-  
vestigations carried out by both the writers and by independent  
experimental investigations (Abiassi 1981; Dalgliesh and Taylor  
1990; Norville et al. 1991).

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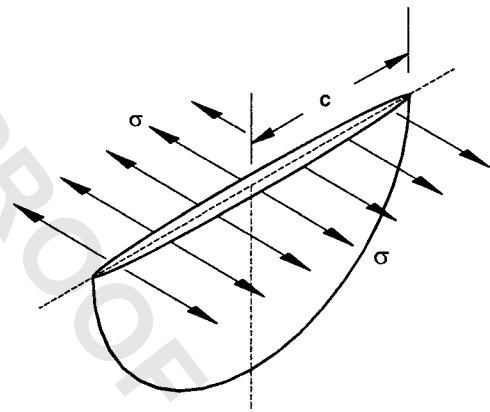


Fig. 1. Half penny crack (adapted from Lawn 1993)

## 75 The Strength of Glass

76 The random molecular structure of glass lacks crystallinity or  
77 long-range order and has no slip planes or dislocations to allow  
78 yield before fracture; consequently, glass exhibits brittle fracture  
79 at a theoretical value between 15,000 and 21,000 MPa (Holloway  
80 1973; Creyke et al. 1982). On freshly drawn glass fibers, tensile  
81 stresses of up to 5,000 MPa have been measured and, even when  
82 incorporated into a resin to form glass reinforced plastic, the glass  
83 fibers have a usable stress of 1,200 MPa (Sedlacek et al. 1995,  
84 Button and Pye 1993). However, the characteristic strength of  
85 architectural glass proposed by the draft European Standard is  
86 45 MPa (CEN 1997) and weathered window glass was reported  
87 to fail at stress levels of around 25 MPa (Button and Pye 1993).  
88 Furthermore, the Institution of Structural Engineers (2000) pro-  
89 posed a value of 8 MPa for the design strength of annealed glass  
90 subjected to long-term stresses.

## 91 Mechanics of Glass Failure

92 These large variations between the theoretical and practical  
93 strength of glass were explained by A. A. Griffith in 1920 (Griffith  
94 1920). Griffith argued that fracture did not start from a pristine  
95 surface, but from preexisting flaws (Griffith flaws) on that sur-  
96 face. Basing his work on the research carried out by Inglis (1913)  
97 on elliptical cavities in plates, Griffith went on to describe crack  
98 growth as a reversible thermodynamic process.

99 Irwin (Lawn 1993) extended the original Griffith energy bal-  
100 ance concept to provide a means of characterizing a material in  
101 terms of its brittleness or fracture toughness. A convenient mate-  
102 rial property defined by Irwin is the stress intensity factor,  $K$ ,  
103 which represents the elastic stress intensity near the crack tip and  
104 depends on the applied loading and the specimen geometry. The  
105 stress intensity factor for mode I loading is  $K_I$ , where mode I  
106 corresponds to normal separation of the crack walls under the  
107 action of tensile stresses and is given by

$$108 \quad K_I = \sigma Y (\pi c)^{1/2} \quad (1)$$

109 where the shape correction factor,  $Y$ , accounts for different ratios  
110 of crack length to specimen width, and the proximity of the crack  
111 to the specimen boundaries. A value of 0.713 has been proposed  
112 for half-penny cracks in a semi-infinite glass specimen shown in  
113 Fig. 1 (Fischer-Cripps and Collins 1995; Lawn 1993). Irwin also  
114 described the resistance to fracture by means of the plane strain  
115 fracture toughness,  $K_{IC}$ , which is the critical value of the stress  
116 intensity factor in Eq. (1), i.e., when  $K_I = K_{IC}$  instantaneous frac-

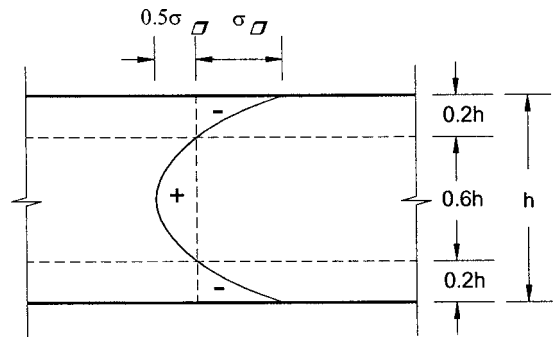


Fig. 2. Parabolic stress distribution resulting from tempering process (adapted from Laufs and Sedlacek 1999)

ture occurs. A typical value for  $K_{IC}$  for soda lime glass is 117  
0.78 MPa<sup>1/2</sup> (Atkins and Mai 1988). 118

119 These fundamentals of fracture mechanics show that the ten-  
120 sile strength of glass is governed by the nature of the surface  
121 flaws and provide an explanation for the large scatter of results  
122 obtained when seemingly identical glass specimens are tested to  
123 failure. The presence of flaws on the glass surface also accounts  
124 for the fact that glass failure can usually be traced back to a single  
125 point of origin, known as the critical flaw, that rarely coincides  
126 with the point of maximum stress. This inherent variability associ-  
127 ated with the surface flaw characteristics implies that the  
128 strength must be treated statistically and that maximum stress-  
129 oriented theory is unable to portray the tensile strength of glass  
130 accurately. 130

131 The most common way of reducing the deleterious effect of  
132 the flaws is by tempering the glass. In this process the glass is  
133 heated and then rapidly quenched, thus introducing a parabolic  
134 stress gradient within the thickness of the glass whereby the  
135 outside surface is stressed in compression (Fig. 2). Any externally  
136 applied force must overcome the surface precompression before  
137 any surface tensile stress can be set up. Tempered glass with a  
138 surface precompression of 120 N/mm<sup>2</sup> is commercially available,  
139 however the presence of other surfaces such as at plate edges,  
140 corners, and holes may distort the parabolic stress distribution and  
141 consequently reduce the surface precompression at these locations  
142 (Laufs and Sedlacek 1999). 142

## Flaw Statistics 143

144 The two-parameter Weibull distribution is reported to provide the  
145 best statistical representation of the strength of glass specimens  
146 (Weibull 1951; Behr et al. 1991). This distribution adopts two  
147 interdependent parameters  $m$  and  $k$  in order to predict the prob-  
148 ability of failure  $P_f$  of a specimen given by Eq. (2) 148

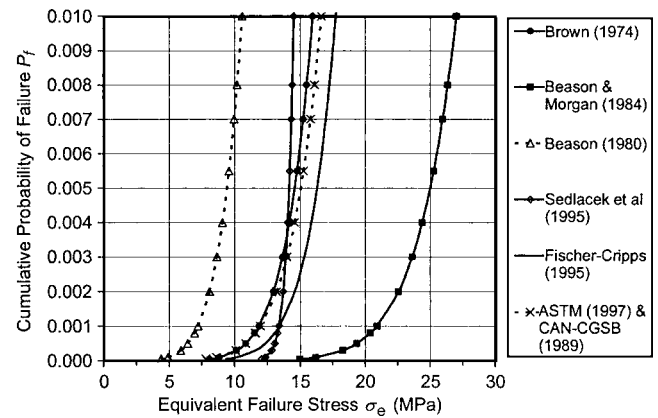
$$P_f = 1 - \exp(-kA\sigma_s^m) \quad (2) \quad 149$$

150 The surface strength parameters  $m$  and  $k$  can only be determined  
151 by experiment and form the basis of the existing failure models  
152 discussed in the ensuing section. From the various numerical and  
153 physical tests carried out (Beason 1980; Beason and Morgan  
154 1984; Sedlacek et al. 1995; Dalgliesh and Taylor 1990; Norvelle  
155 et al. 1991, and Charles 1958), it may be concluded that the  
156 long-term strength of glass depends on the following parameters: 156

- 157 1. Duration of application of load. 157
- 158 2. Surface area of glass exposed to the tensile stress. 158
- 159 3. Orientations of the surface flaws with respect to the principle  
160 stresses on the surface of the glass. 160

**Table 1.** Surface Strength Parameters

Failure model	Surface strength parameters	
	$m$	$k$
Brown (1974); Fischer-Cripps and Collins (1995) As-received glass	7.3	$5.1 \times 10^{-57} \text{ m}^{-2} \text{ Pa}^{-7.3}$
Beason (1980) Weathered glass	6	$7.19 \times 10^{-45} \text{ m}^{-2} \text{ Pa}^{-6}$
Beason and Morgan (1984) As-received glass	9	$1.32 \times 10^{-69} \text{ m}^{-2} \text{ Pa}^{-9}$
ASTM (1997); CAN-CGSB (1989) Weathered glass	7	$2.86 \times 10^{-53} \text{ m}^{-2} \text{ Pa}^{-7}$
Sedlacek et al. (1995); CEN/TC129/WG8 (CEN 1997) As-received glass	25	$2.35 \times 10^{-188} \text{ m}^{-2} \text{ Pa}^{-25}$



**Fig. 3.** PDF at low probabilities of failure (dotted lines represent weathered glass)

- 161 4. Environmental conditions, especially humidity.
- 162 5. Magnitude and distribution of the net surface tensile stresses.

**163 Existing Failure Models**

164 The increasing use of glass as a load-bearing material has led to  
 165 the development of a number of failure prediction models. The  
 166 aim of these failure models is to arrive at a value of allowable  
 167 load or stress for an acceptable probability of failure in terms of  
 168 the environmental and geometrical parameters. The earliest such  
 169 failure model, the *load duration theory*, was proposed by Brown  
 170 in 1974 (Brown 1974). Beason (1980) and Beason and Morgan  
 171 (1984) developed the *glass failure prediction model*, which con-  
 172 stitutes the backbone of the United States (ASTM 1997) and  
 173 Canadian (CAN/CGSB 1989) standards, and is based on the  
 174 semiempirical thermodynamic formulations of Charles (1958).  
 175 Recently an alternative treatment of the failure of brittle solids,  
 176 derived from elastic fracture mechanics and subcritical crack  
 177 growth, has emerged in the form of the crack growth models of  
 178 Sedlacek et al. (1995) and Fischer-Cripps and Collins (1995).  
 179 More recently, Porter and Houlsby (2001) have proposed an al-  
 180 ternative design method with underlying fracture mechanics for-  
 181 mulations similar to those adopted by Fischer-Cripps and Collins.  
 182 These failure prediction models constitute a valuable source of  
 183 information for the structural design of glass. However, these  
 184 models have never been compared and therefore it seems oppor-  
 185 tune to do so.

186 The main discrepancies between the existing failure models  
 187 arise from the adoption of dissimilar surface strength parameters  
 188  $m$  and  $k$  and from the different representations of the load  
 189 duration, surface area, and flaw orientation effects on the tensile  
 190 strength of glass. These aspects are discussed in detail in the  
 191 following sequel.

192 The surface strength parameters shown in Table 1 and the  
 193 probability distribution functions (PDFs) plotted in Fig. 3 reveal  
 194 that there is good agreement between the functions adopted by  
 195 Brown (1974), Fischer-Cripps and Collins (1995), ASTM (1997),  
 196 and CAN/CGSB (1989). The Sedlacek et al. (1995) PDF, which  
 197 forms the basis of the draft European Standard (CEN 1997),  
 198 shows reasonable agreement with the Brown (1974), Fisher-  
 199 Cripps & Collins (1995), ASTM (1997), and CAN/CGSB (1989)  
 200 functions at low probabilities of failure. There is however a large  
 201 disparity between the Sedlacek et al. (1995) model and the others  
 202 at higher probabilities of failure. This is due to a high surface

strength parameter ( $m=25$ ) adopted by this model indicating an  
 uncharacteristically low variability in glass strength. It is impor-  
 tant to note that the Beason and Morgan (1984) PDF was derived  
 from testing weathered glass and consequently provides the low-  
 est strength values. The ASTM E-1300-97 (ASTM 1997) and  
 CAN/CGSB 12.20-M89 (CAN/CSB 1989) functions were also  
 formulated for weathered glass, however they provide a more  
 optimistic strength prediction than Beason and Morgan (1984).

210 These differences in surface strength parameters result in con-  
 211 siderable differences in the strength values of glass in practical  
 212 applications. An example of this is shown in Table 2 in which the  
 213 failure stresses have been derived for a 1 m<sup>2</sup> plate of annealed  
 214 glass with a uniformly applied surface tensile stress and a 60 s  
 215 load duration.

216 The existing models also account for degradation of the tensile  
 217 strength of glass with increasing load duration. This phenomenon,  
 218 termed stress corrosion (or static fatigue), is caused by the sub-  
 219 critical crack growth on the glass surface at stress levels below  
 220 instantaneous failure stress. Under constant load and constant  
 221 relative humidity, the 60 s equivalent stress may be expressed by  
 222

$$\sigma_e = \sigma_s \left( \frac{t_f}{60} \right)^{1/n} \quad (3)$$

223 where  $\sigma_s$  is derived from Eq. (2) and  $n$ =stress corrosion constant,  
 224 the magnitude of which is dependent on environmental condi-  
 225 tions, especially humidity.

226 The stress corrosion effects adopted by the various models, for  
 227 a typical 1 m<sup>2</sup> uniformly stressed plate and a probability of failure  
 228

**Table 2.** Comparison of 1960s Equivalent Stresses

Failure model	1960s Equivalent failure stress (MPa)	
	$P_f=1/125$	$P_f=1/1,000$
Brown (1974)	15.50	11.85
Beason (1980)	10.19	7.20
Beason and Morgan (1984)	26.24	20.89
ASTM (1997); CAN-CGSB (1989)	16.11	11.96
Sedlacek et al. (1995), CEN/TC129/WG8 (CEN 1997)	14.39	13.38
Fischer-Cripps and Collins (1995)	17.25	13.41



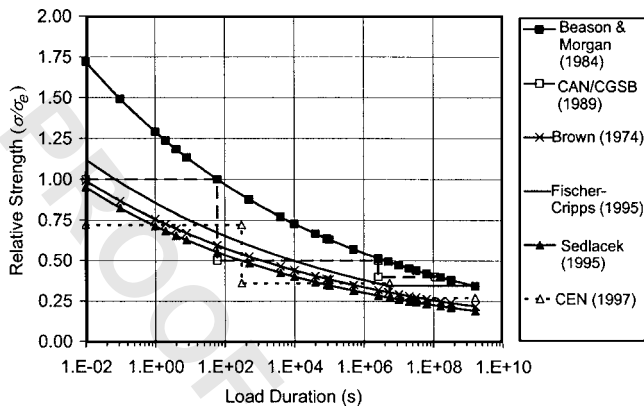


Fig. 4. Stress corrosion curves for annealed glass

of 1/1,000, are shown in Fig. 4. The asymptotical ends to the Fisher-Cripps and Collins and the Sedlacek et al. load duration curves represent the static fatigue limits beyond which subcritical crack growth will not occur. The static fatigue limit shown on the Sedlacek et al. (1995) curve is that adopted by the draft European Standard (CEN 1997).

From Fig. 4, it is evident that the Beason and Morgan (1984) model provides the most optimistic prediction of glass strength. This is a direct result of the surface strength parameters adopted, as discussed in the previous section. However, the models represented by the continuous functions are in close agreement in terms of the relative strength of glass between long-term and short-term loads. For example, the tensile strength of glass subjected to a constant load for a 1-month duration ranges between 0.49 and 0.53 of the 60 s strength depending on the failure model adopted.

The CAN/CGSB (1989) step function provides a good lower-bound approximation to the Brown (1974) curve. The Fischer-Cripps and Collins (1995) and Sedlacek et al. (1995) load duration curves are in very close agreement. However there are two anomalies in the draft European Standard (CEN 1997). The first is that the step function set out by this standard straddles the Sedlacek et al. curve and therefore does not appear to provide a safe representation of the stress corrosion characteristics proposed by Sedlacek et al. (1995). Second, the static fatigue ratio of 27% adopted by the European standard is outside the 32–38% range of static fatigue limits reported elsewhere (Wan et al. 1961; Shand 1965; Wiederhorn and Bolz 1970; Wiederhorn 1977; Michalske 1983). In this latter case the European standard seems to overestimate the deleterious effects of stress corrosion.

The existing failure models also account for the reduction in the tensile strength of glass with increasing surface area. The relationships between the strength and the stressed surface are shown in Fig. 5. The relative strength on the ordinate Y axis represents the ratio of the tensile strength for a given surface area to the tensile strength of a 1 m<sup>2</sup> glass plate and equates to  $(A_o/A)^{1/m}$ .

Fig. 5 reveals that there is good agreement between the Fischer-Cripps and Collins (1995) curve, which is identical to Brown's relationship, and the ASTM (1997) and CAN/CGSB (1989) curves. There is less agreement between the above-mentioned curves and the relationship proposed by Beason and Morgan (1984), but the differences are within ±3% for a surface area between 0.5 and 4 m<sup>2</sup>. The Sedlacek et al. (1995) strength versus area relationship implies that the surface area has a less pronounced effect on the strength of glass than that proposed by

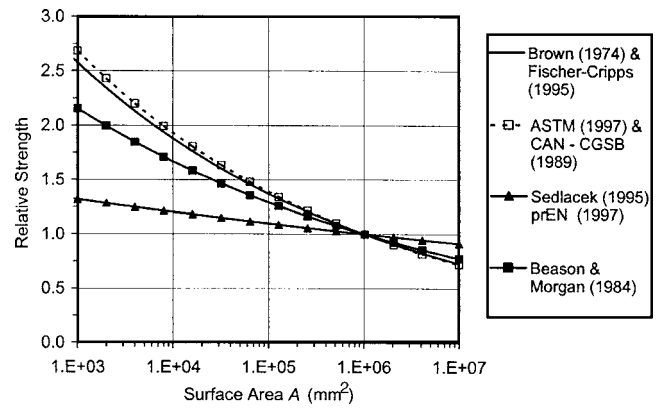


Fig. 5. Relative strength of annealed glass with variations in surface area (dotted lines represent weathered glass)

the other failure models. This discrepancy is mainly attributed to the relatively high surface strength parameter ( $m=25$ ) adopted by the European standard.

Surprisingly, the crack growth models proposed by Fischer-Cripps and Collins (1995) and Sedlacek et al. (1995) fail to consider the effect of the orientation of the surface flaws and the magnitude of principle stresses on the tensile strength of glass. This is clearly a shortcoming, particularly because the propensity of a flaw to initiate failure (and hence the tensile strength) is a function of the orientation of the flaw with respect to the major and minor principal stresses  $\sigma_{max}$  and  $\sigma_{min}$  (Fig. 6).

The variations of normal stresses with flaw orientations are included in the glass failure prediction model proposed by Beason (1980) which was subsequently extended by Beason and Morgan (1984). This was achieved by introducing a biaxial stress modification factor based on Weibull's formulation (Weibull 1951) such that Eq. (2) may be rewritten as

$$P_f = 1 - \exp[kA(c_b\sigma_s)^m] \quad (4)$$

where

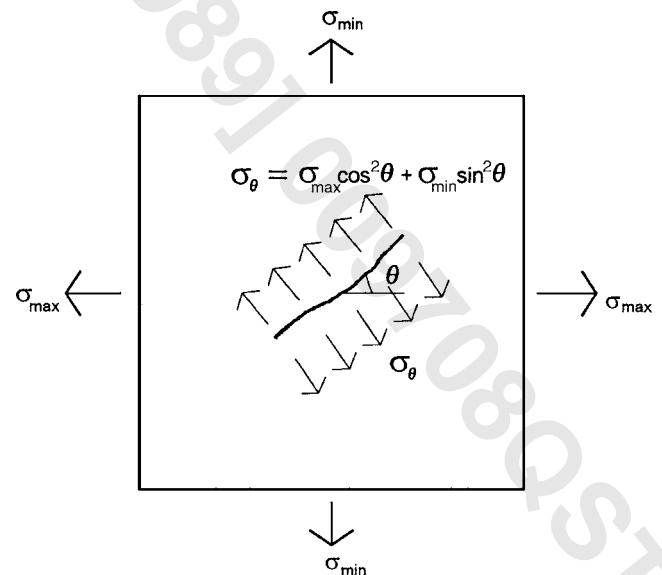


Fig. 6. Flaw orientation

$$c_b = \left[ \frac{2}{\pi} \int_0^\alpha (\cos^2 \theta + n \sin^2 \theta)^m d\theta \right]^{1/m} \quad (5)$$

## 295 Formulation of General Crack Growth Model

296 From the comparisons carried out in the previous sections it is  
 297 evident that the crack growth models are the closest representa-  
 298 tion of the failure mechanism at microscopic level, and are there-  
 299 fore, best suited to determine the probability of failure of glass.  
 300 However, these models do not take into account all five factors  
 301 listed in the previous section that are known to affect glass  
 302 strength. Notably, the crack growth models proposed by Sedlacek  
 303 et al. (1995) and Fischer-Cripps & Collins (1995) do not consider  
 304 the effects of surface flaw orientation on the tensile strength of  
 305 glass. From these two existing failure models the Fischer-Cripps  
 306 and Collins model is preferred by the writers as it adopts a veri-  
 307 fied stress corrosion limit and surface strength parameters that are  
 308 within the range of values reported elsewhere (Table 1).

### 309 Tensile Strength of Glass

310 Since tempered glass is used in the majority of structural glass  
 311 applications it is therefore essential to account for the surface  
 312 precompression in the proposed model. In tempered glass, the  
 313 stress corrosion phenomenon only occurs after the applied tensile  
 314 surface stress exceeds the residual precompression  $\sigma_r$ . Eq. (1)  
 315 may therefore be used to characterize the instantaneous failure  $\sigma_s$   
 316 of glass with a modification,  $\sigma_r$ , to take into account the surface  
 317 precompression induced by the tempering process as given by  
 318 Eq. (6).

$$319 \quad \sigma_s = [K_{IC}/Y(\pi c)^{1/2}] + \sigma_r \quad (6)$$

320 It is important to note that the thermally induced surface precom-  
 321 pression is distorted close to free edges and holes in the glass  
 322 (Laufs and Sedlacek 1999). Therefore the magnitude of  $\sigma_r$  de-  
 323 pends on the location under consideration.

324 If the applied tensile stress exceeds the thermally induced sur-  
 325 face precompression  $\sigma_r$  at a specific location, the net tensile stress  
 326 may either cause a critical flaw to fail instantaneously or it may  
 327 cause a flaw to grow subcritically, under sustained load, until it  
 328 reaches a length that will cause failure of the glass. Since the  
 329 static and pseudostatic load durations encountered in practice nor-  
 330 mally include medium and long-term loads ranging from a few  
 331 minutes to the full service life, it is necessary to transform the  
 332 instantaneous annealed glass failure strength,  $\sigma_s$ , to an equivalent  
 333 (same probability) failure strength,  $\sigma_f$ , by taking into account the  
 334 stress corrosion characteristics of glass. The  $\sigma_f/\sigma_s$  relationship,  
 335 termed the stress corrosion modification factor  $k_{mod}$ , is equivalent  
 336 to the ratio of stress concentration factors  $K_{IC}/K_I$  defined by  
 337 Fischer-Cripps and Collins (1995), such that

$$338 \quad K_I = \frac{K_{IC}\sigma_s}{\sigma_f} \quad (7)$$

339 The time required for a flaw to grow subcritically from its initial  
 340 unstressed size to a final critical size that will cause failure was  
 341 also derived by Fischer-Cripps and Collins (1995) and is given by

$$342 \quad t_f = \frac{2K_I^{2-n}}{D(n-2)\sigma_s^2 Y^2 \pi} \quad (8)$$

343 by substituting Eq. (4) into Eq. (5) gives

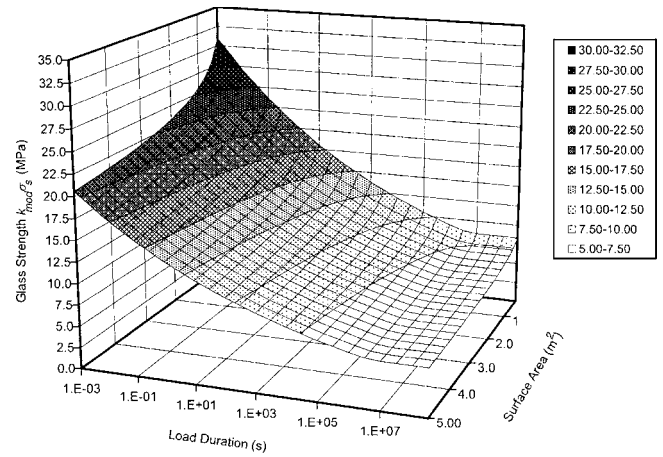


Fig. 7. Failure envelope for annealed glass

$$k_{mod} = \frac{\sigma_f}{\sigma_s} = \left[ \frac{t_f D (n-2) K_{IC}^{n-2} \sigma_s^2 Y^2 \pi}{2} \right]^{1/(n-2)} \geq 0.346 \quad (9) \quad 344$$

where  $K_{IC}$ =critical stress intensity factor with a value of  $0.78 \times 10^6 \text{ m}^{1/2} \text{ Pa}$  (Atkins and Mai 1988),  $Y$ =shape correction factor with a value of 0.713 for half-penny cracks (Atkins and Mai 1988),  $n$ =static fatigue constant taken as 16, and  $\sigma_s$ =instantaneous stress applied for a time  $t_f$ . However, the stress corrosion modification factor,  $k_{mod}$ , is limited by a stress (or crack length) below which subcritical crack growth will not occur. This is represented by the 0.346 limit in Eq. (9).

The surface tensile strength  $\sigma_f$  may therefore be obtained from Eq. (10) as suggested in the Draft European Standard (CEN 1997):

$$\sigma_f = k_{mod}\sigma_s + \sigma_r/\gamma_v \quad (10) \quad 356$$

where  $k_{mod}$ =stress corrosion modification factor obtained from Eq. (9),  $\sigma_s$ =instantaneous failure stress for annealed glass obtained from Eq. (2) for a required probability of failure  $P_f$ ,  $\sigma_r$ =surface precompression due to the tempering process provided by the manufacturer, and  $\gamma_v$ =safety factor depending on the surface precompression, the magnitude of which depends on the level of quality assurance. The draft European Standard (CEN 1997) uses the material safety factor,  $\gamma_v$ , to account for both the level of quality assurance and the reduced magnitude of precompression close to the edges of the glass. The values of  $\gamma_v$  suggested by the draft European Standard range between 1.5 and 2.4.

The combined influence of load duration and stressed area on the strength of glass may be expressed by means of a failure envelope for a given probability of failure (Fig. 7). The surface plotted in Fig. 7 represents the failure stresses for a range of load durations  $t_f$  and surface areas  $A$ , and a probability of failure  $P_f$  of 1/1,000. A glass specimen with a known surface area and load duration may be represented by a point in  $P_f/A/t_f$  space. A point above the surface indicates that the probability of failure is greater than 1/1,000 and a point on or below the surface indicates that the glass specimen in question has a probability of failure that is equal to or less than 1/1,000, respectively. Interestingly, the plan view of this failure surface shown in Fig. 8 is very convenient for obtaining graphically the  $k_{mod}\sigma_s$  term used in Eq. (5).

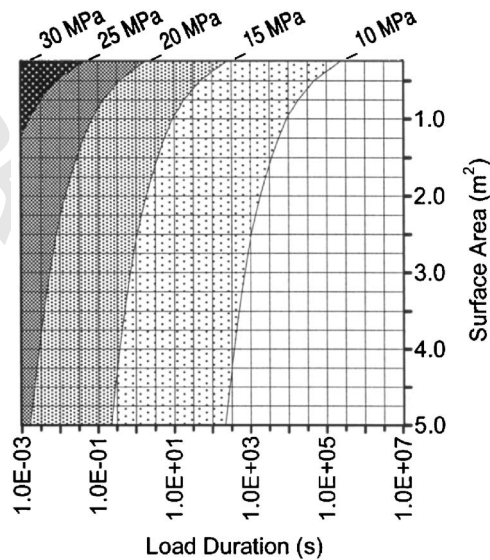


Fig. 8. Plan view of failure envelope

### 381 Applied Stress

382 The risk of failure experienced by a glass plate is related to the  
 383 magnitude of the applied stresses which act normal to the longi-  
 384 tudinal axis of the surface flaws as shown in Fig. 6. This is also  
 385 known as mode I loading. The term  $\sigma_\sigma$  in Eqs. (2) and (9) are  
 386 both derived from experiments where the stress is applied perpen-  
 387 dicular to a crack (Wan et al. 1961; Shand 1965; Wiederhorn and  
 388 Bolz 1970; Wiederhorn 1977; Michalske 1983). Although it is  
 389 unlikely that the precise orientation of surface flaws on glass  
 390 plates will ever be known, it is nevertheless possible to incorpo-  
 391 rate the variation of normal stress with flaw orientation as  
 392 suggested by Beason and Morgan (1984) based on the formula-  
 393 tions of Weibull (1951) as shown in Eqs. (4) and (5).  
 394 If the critical flaw is oriented at  $\theta$  from the plane of the maxi-  
 395 mum principal stress  $\sigma_{\max}$  and assuming that mode I loading is  
 396 the only contributor to crack propagation, the stress applied per-  
 397 pendicular to a flaw shown in Fig. 6 may be rewritten as

$$398 \quad \sigma_a = c_b \sigma_{\max} \quad (11)$$

399 where  $c_b$ =biaxial stress correction factor ranging from 0.77, for  
 400  $\sigma_{\min}/\sigma_{\max}=1.0$ , to unity, for  $\sigma_{\min}/\sigma_{\max}=-1.0$  (Beason and  
 401 Morgan 1984).

402 This relationship between the normal principle stresses and the  
 403 probability of failure for a 1 m<sup>2</sup> glass plate is plotted in Fig. 9.

404 The summation of all the stresses present on the glass surface  
 405 may be conservatively taken as the maximum applied stress  $\sigma_a$   
 406 over the whole plate surface. This is usually overly conservative  
 407 and a more accurate approach is to summate the contributions of  
 408 various stressed areas on the surface of the glass to the probability  
 409 of failure. This approach may be derived from the original formu-  
 410 lations of Beason and Morgan (1984) and the elaborations in the  
 411 draft European Standard (CEN 1997), Overend et al. (1999), and  
 412 Overend (2002) such that

$$413 \quad \sigma_p = \left[ \frac{1}{A} \int_{\text{area}} (c_b \sigma_{\max})^m dA \right]^{1/m} \quad (12)$$

414 where  $\sigma_p$ =equivalent uniform stress and represents a weighted  
 415 average of the surface tensile stress distribution on the glass plate.  
 416 This equation is very convenient because it transforms the actual

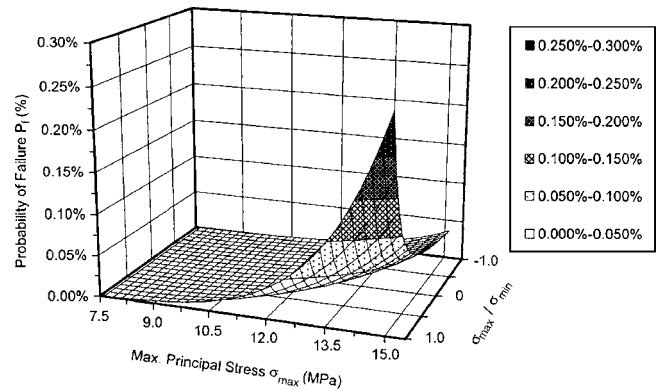


Fig. 9. Variations in probability of failure with nonuniform biaxial stresses

and complex stress distribution on the glass surface to a single  
 equivalent stress. 417  
 418

### Structural Adequacy and Design 419

The effective uniformly applied stress  $\sigma_p$  derived from Eq. (12) 420  
 may be finally compared to the failure strength  $\sigma_f$  from Eq. (10) 421  
 to ensure that 422

$$\sigma_f \geq \sigma_p \quad (13) \quad 423$$

The accuracy of this method clearly depends on the ability to 424  
 execute Eq. (12) i.e., to subdivide the glass surface into areas of 425  
 equal stress, and to subsequently summate the contribution of 426  
 these areas. This procedure makes the proposed design method 427  
 unattractive for manual computation. A computer algorithm was 428  
 therefore developed to determine the equivalent uniform stress  $\sigma_p$  429  
 automatically. 430

The algorithm is written in Visual Basic computer language 431  
 and makes use of the results obtained from LUSAS, a commer- 432  
 cially available finite-element (FE) analysis software (FEA 1999). 433  
 Input to the algorithm consists of the coordinates of the surfaces 434  
 to be analyzed and the magnitude of the surface precompression 435  
 due to the tempering process. The algorithm may be used with a 436  
 number of commonly used elements ranging from three-noded 437  
 triangular elements to 20-noded brick elements and conveniently 438  
 calculates the surface areas,  $dA$ , and averaged principal tensile 439  
 stress,  $\sigma_{\max}$ , for each element of the FE model. All the elements 440  
 subjected to a compressive stress are eliminated from this sum- 441  
 mation. The algorithm uses this data to automatically compute the 442  
 equivalent uniform stress,  $\sigma_p$ , for the whole surface from Eq. (11). 443  
 The algorithm also creates a spreadsheet containing a listing of 444  
 these calculations and a summary of the results for the entire 445  
 surface. 446

### Verification of Failure Models 447

Experimental investigations were undertaken to verify the accu- 448  
 racy of the proposed failure model and to compare it to the failure 449  
 predictions obtained from existing failure models. The experi- 450  
 mental investigations consisted of undertaking a series of: 451

1. Ring-on-ring (co-axial ring) tests. 452
  2. Laterally loaded, simply supported plate tests. 453
- These two setups were selected as they provide diverse stress 454  
 distributions ranging from relatively concentrated stresses in the 455



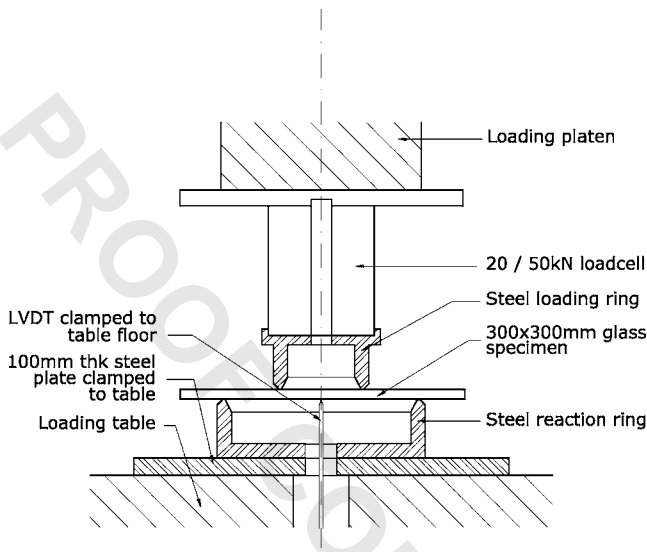


Fig. 10. Ring-on-ring test setup

ring-on-ring tests, to the shallow-gradient stresses that are typical of simply supported plates subjected to uniform lateral loading.

#### 458 Ring-on-Ring Investigations

A series of ring-on-ring tests were carried out by the writers on as-received glass specimens measuring  $300 \times 300 \times 6$  mm thick. The tests were performed by placing the glass plate on a circular steel reaction ring and applying on its opposite surface a load transmitted through a steel loading ring, until failure occurs (Fig. 10). The purpose of this test is to achieve a uniform tensile stress field that is independent of edge effects.

In all 49 ring-on-ring tests were successfully performed. These were composed of 30 annealed glass specimens and 19 tempered glass specimens to BS 6206 class A (BSI 1981) and tested by means of a 51-mm-diameter steel loading ring and one of three different steel reaction rings with 65, 127, and 200 mm diameters (Fig. 10).

Geometrically nonlinear finite-element analysis of the ring-on-ring specimens was undertaken to provide an accurate representation of the surface stresses for the expected large deflections. The maximum surface stresses and maximum deflections obtained from the finite-element analyses were within  $\pm 3\%$  of those obtained from the experimental investigation. Furthermore, the FE analysis confirmed that the stress concentrations beneath the loading ring were within  $\pm 2\%$  of the stress at midspan. Surface stresses obtained from the finite-element analysis were transformed to an equivalent uniform failure stress and an associated probability of failure by the afore-mentioned computer algorithm. Furthermore, since most failures occurred within the loading ring area, a biaxial stress correction factor  $c_b = 1$  was used in Eq. (12).

The probability distribution functions obtained from the proposed GCGM and computer algorithm shows good agreement with the annealed and tempered glass ring-on-ring test data at mean and low probabilities of failure ( $\pm 4\%$  for both  $P_f = 0.5$  and  $P_f = 0.1$ ). However, the predictions for high probabilities of failure are less accurate particularly for the tempered glass where the variation at  $P_f = 0.9$  is greater than 12%. (Fig. 11). The increasing inaccuracy with increasing probability of failure for both annealed and tempered glass specimens cannot be fully explained, however the more pronounced variability witnessed in the

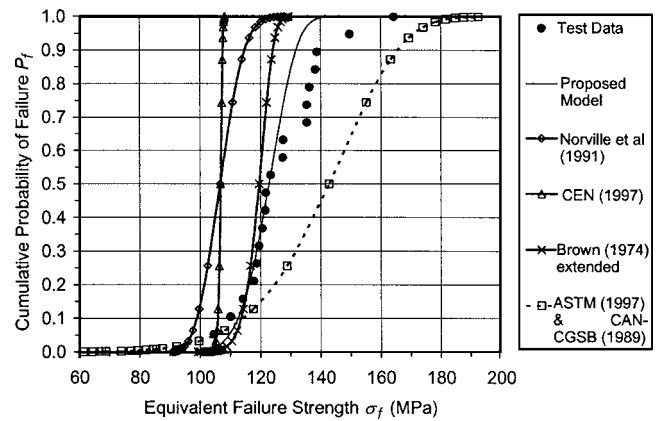


Fig. 11. Test results and failure models for ring-on-ring tempered glass (dotted lines represent weathered glass)

strength of the tempered glass specimens may be attributed to the additional variations caused by the tempering process.

#### Laterally Loaded Plate Investigations

Two sets of published annealed glass failure data (Beason 1980; Dalglish and Taylor 1990) and one set of published failure data for tempered glass (Norville et al. 1991) were used to compare the performance of the existing and the proposed failure models and to further test the validity of the proposed GCGM together with the computer algorithm.

Convergence testing of the FE model ensured that the predicted stresses and deflections were within  $\pm 1\%$  of those reported from the experimental investigations.

The computer algorithm was used to summate the major principal surface stresses obtained from each load increment of the FE analysis. The surface precompression  $\sigma_r$  was set to zero for the annealed glass specimens and to  $69 \text{ N/mm}^2$  for the tempered glass specimens. This value corresponds to the surface stress measurements carried out by Norville et al. (1991) and agrees with the minimum required value specified in ASTM E-1300 (ASTM 1997). The resulting equivalent uniform stress,  $\sigma_p$ , obtained from the algorithm, was used to determine the probability of failure  $P_f$  at every load increment by using Eq. (2).

The relationships between the uniformly distributed, 60 s equivalent load  $P_{60}$  and the probability of failure  $P_f$  were plotted in Figs. 12 and 13. Fig. 12 shows the probability of failure versus

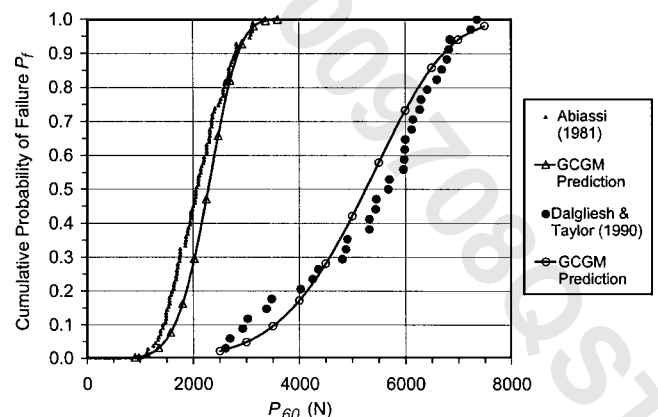


Fig. 12. Test results and GCGM predictions for annealed glass

520 the 60 s failure load of the published test results (Beason 1980;  
 521 Dalgliesh and Taylor 1990) and the corresponding GCGM predic-  
 522 tions for uniformly loaded, simply supported rectangular annealed  
 523 glass. Fig. 13 shows the probability of failure versus the 60 s  
 524 failure load of the published test results (Norville et al. 1991) and  
 525 the corresponding GCGM predictions for uniformly loaded, sim-  
 526 ply supported rectangular tempered glass.

527 The predicted relationship between the 60 s equivalent loads,  
 528  $P_{60}$ , and the probability of failure  $P_f$  (Fig. 13) are in good agree-  
 529 ment with the annealed and tempered glass test results. Table 3  
 530 provides a quantitative comparison of the proposed GCGM at the  
 531 low probabilities of failure generally used in glass design prac-  
 532 tice. This table also indicates how the GCGM compares with  
 533 other failure prediction models at low probabilities of failure. All  
 534 probabilities of failure in this table have been computed by using  
 535 the respective surface strength parameters  $m$  and  $k$  from Table 2.

536 From Table 3 it is evident that all failure prediction models  
 537 provide a more accurate, albeit sometimes unsafe, representation  
 538 of glass strength when compared to the results obtained from the  
 539 simpler maximum stress approach. The GCGM appears to predict  
 540 the probability of failure more closely than the other models for  
 541 both the Dalgliesh and Taylor annealed glass (Dalgliesh and Taylor  
 542 1990) and the Norville et al. tempered glass (Norville et al. 1991).  
 543 The ASTM E-1300 model provides the best predictions of Beason  
 544 annealed glass tests (Beason 1980). However, it is important to  
 545 note that the ASTM E-1300 was partially derived from the  
 546 Beason (1980) tests. The draft European Standard (CEN 1997),  
 547 derived from the Sedlacek et al. (1995) model, adopts an unchar-  
 548 acteristically high surface strength parameter,  $m$ , and a low pa-  
 549 rameter  $k$ . This in effect restricts its use to very low probabilities  
 550 of failure ( $<8/1,000$ ) as observed in the preceding sections of  
 551 this paper. This may partly explain the poor predictions at rela-  
 552 tively high probabilities of failure shown in Table 3.

### 553 Conclusions

554 A number of existing glass failure models that are used in glass  
 555 design are reviewed, and the discrepancies between these models,  
 556 particularly the interpretation of load duration, surface area, and  
 557 stress distribution, have been identified. Substantial differences  
 558 have been noted in the magnitude of surface strength parameters  
 559 and in the effects of the relative magnitude of major and minor  
 560 principal stress acting on the surface of the glass. These variations

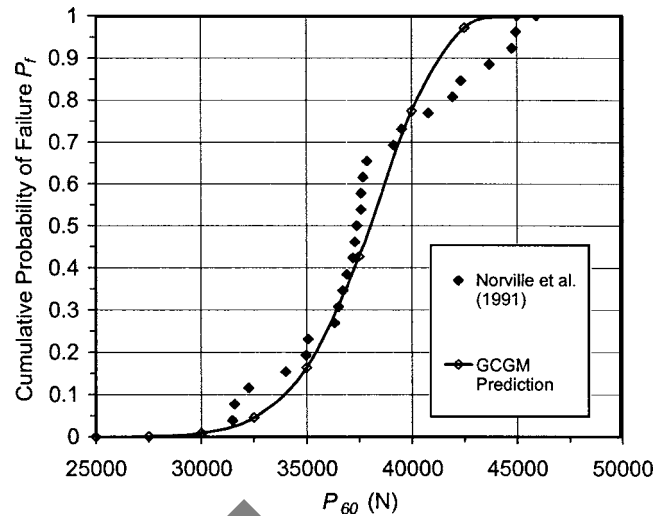


Fig. 13. Test results and GCGM predictions for tempered glass

are shown to translate into considerable differences in the prob- 561  
 abilities of failure obtained from the respective models. 562

A GCGM is proposed by extending the formulations of 563  
 Fischer-Cripps and Collins (1995). The GCGM combines statisti- 564  
 cal theory with linear elastic fracture mechanics and allows the 565  
 surface tensile strength of both annealed and tempered glass to be 566  
 determined graphically. Furthermore, since the proposed and ex- 567  
 isting failure models are unattractive for manual computation, a 568  
 computer algorithm is also put forward. This algorithm calculates 569  
 the equivalent uniform stress automatically from the results of the 570  
 finite-element analysis performed on the glass element. The pre- 571  
 dictions obtained by applying the proposed GCGM are in good 572  
 agreement with the strength values obtained from ring-on-ring 573  
 experimental investigations carried out by the writers and experi- 574  
 mental investigations on laterally loaded rectangular glass plates 575  
 carried out by other researchers. Furthermore the use of the com- 576  
 puter algorithm resulted in a substantial reduction in computation 577  
 time. 578

Further validation of the proposed GCGM and design method- 579  
 ology is required before it can be used by the general engineering 580  
 community. This includes fundamental research on the nature of 581  
 flaws in glass and the mechanics of glass failure that will serve to 582  
 verify the constants such as the stress intensity factor and the 583

Table 3. Comparison of Reported Test Results and Predicted Probability of Failure

Source	Arbitrary load (kN)	Reported	Probability of failure			
			Predicted from proposed GCGM <sup>a</sup>	Predicted from ASTM & CAN/CGSB <sup>b</sup>	Predicted from CEN <sup>c</sup>	Predicted using maximum stress
Annealed glass Beason (1980)	1.125	0.018	0.010	0.015	0.00026	0.087
Annealed glass Dalgliesh and Taylor (1990)	2.600	0.026	0.025	0.039	0.001	0.094
Tempered glass Norville et al. (1991)	31.500	0.039	0.031	0.001	0.053	1.000

<sup>a</sup>Probability of failure calculated using FE analysis and the computer algorithm with Fischer-Cripps and Collins (1995) surface strength parameters from Table 2.

<sup>b</sup>Probability of failure calculated using FE analysis and computer algorithm with the ASTM (1997) and CAN/CGSB (1989) surface strength parameters shown in Table 2.

<sup>c</sup>Probability of failure calculated using FE analysis and computer algorithm with the CEN (1997) surface strength parameters shown in Table 2.



584 stress corrosion limit in glass; a full reliability analysis of the test  
 585 data used to calibrate the United States and European standards in  
 586 order to verify the surface strength parameters, and the extension  
 587 of the GCGM to buckling instability and impact loads.  
 588 Current and planned research at the University of Nottingham  
 589 in collaboration with other research institution in the United  
 590 Kingdom and in Europe includes the application of the GCGM to  
 591 plates in compression and to built-up glass elements; the testing  
 592 of a series of 40-year old weathered glass specimens; the use of  
 593 the GCGM to optimize bolted and adhesive connections in glass;  
 594 and investigations on the postbreakage performance of safety  
 595 glass. The results from these research projects will seek to verify  
 596 the proposed GCGM design methodology to a wider range of  
 597 structural glass elements.

## 598 Notation

599 *The following symbols are used in this paper:*

600  $A$  = surface area;  
 602  $A_o$  = datum surface area ( $=1 \text{ m}^2$ );  
 603  $c$  = flaw length;  
 604  $c_b$  = biaxial stress correction factor;  
 605  $D$  = material fracture constant;  
 606  $h$  = plate thickness;  
 607  $K_I$  = stress intensity factor for mode I loading;  
 608  $K_{IC}$  = critical stress intensity factor for mode I  
 609 loading (plane strain fracture toughness);  
 610  $k$  = surface strength parameter;  
 611  $k_{\text{mod}}$  = stress corrosion ratio;  
 612  $m$  = surface strength parameter, Fourier series  
 613 numerical factor;  
 614  $n$  = static fatigue constant;  
 615  $P_f$  = probability of failure;  
 616  $P_{60}$  = 60 second equivalent failure load;  
 617  $t_f$  = load duration;  
 618  $Y$  = shape factor;  
 619  $\gamma_v$  = material safety factor for tempered glass;  
 620  $\sigma$  = general applied stress;  
 621  $\sigma_a$  = stress applied perpendicular to crack;  
 622  $\sigma_e$  = 60 s equivalent failure stress;  
 623  $\sigma_f$  = surface tensile strength of glass;  
 624  $\sigma_{\text{max}}$  = major principal stress;  
 625  $\sigma_{\text{min}}$  = minor principal stress;  
 626  $\sigma_p$  = equivalent uniform stress;  
 627  $\sigma_r$  = surface precompression due to tempering  
 628 process; and  
 629  $\sigma_s$  = instantaneous failure stress.

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