

Finite Element Formulation for Shells - Handout 5 -

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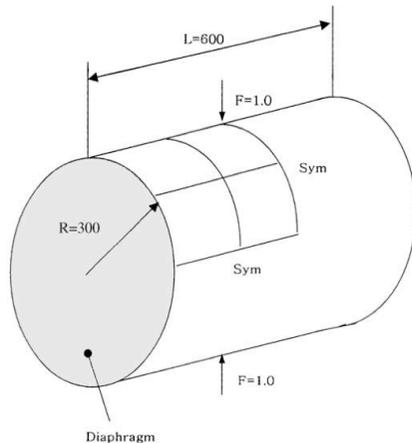
Completed Version

Overview of Shell Finite Elements

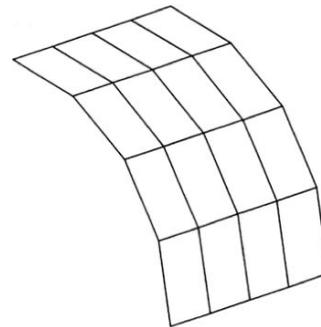
- There are three different approaches for deriving shell finite elements
 - Flat shell elements
 - The geometry of a shell is approximated with flat finite elements
 - Flat shell elements are obtained by combining plate elements with plate stress elements
 - “Degenerated” shell elements
 - Elements are derived by “degenerating” a three dimensional solid finite element into a shell surface element

Flat Shell Finite Elements

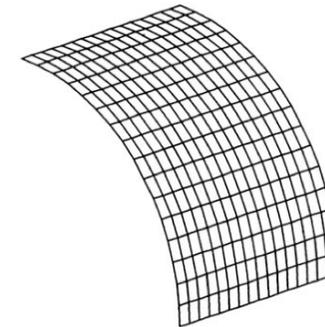
- Example: Discretization of a cylindrical shell with flat shell finite elements



Cylindrical shell



Coarse mesh

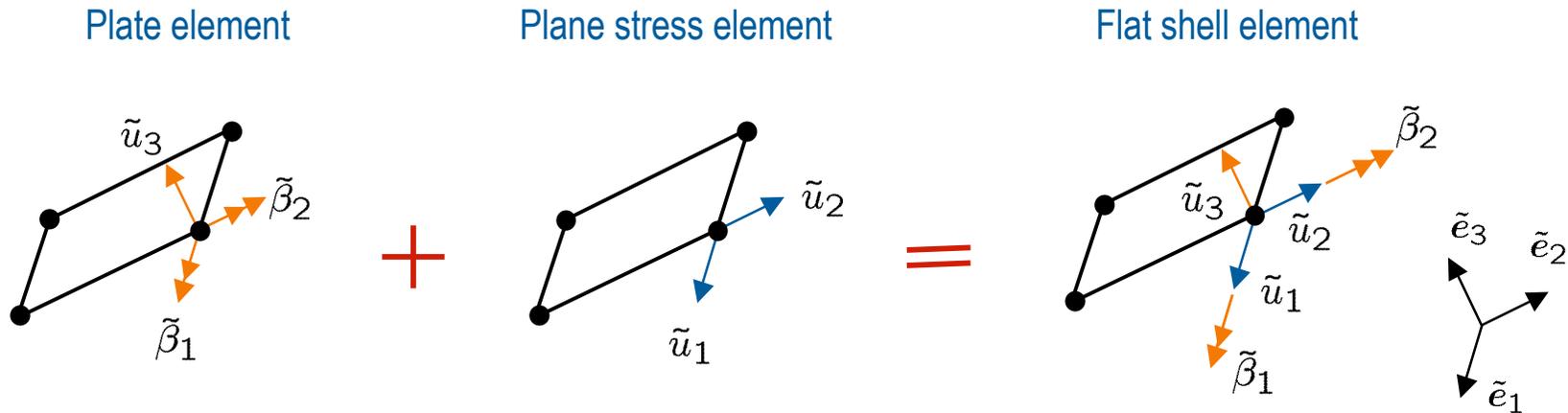


Fine mesh

- Note that due to symmetry only one eighth of the shell is discretized
- The quality of the surface approximation improves if more and more flat elements are used
- Flat shell finite elements are derived by superposition of plate finite elements with plane stress finite elements
 - As plate finite elements usually Reissner-Mindlin plate elements are used
 - As plane stress elements the finite elements derived in 3D7 are used
- Overall approach equivalent to deriving frame finite elements by superposition of beam and truss finite elements

Four-Noded Flat Shell Element -1-

- First the degrees of freedom of a plate and plane-stress finite element in a local element-aligned coordinate system are considered



- The local base vectors \tilde{e}_1 and \tilde{e}_2 are in the plane of the element and \tilde{e}_3 is orthogonal to the element
- The plate element has three degrees of freedom per node (one out-of-plane displacement and two rotations)
- The plane stress element has two degrees of freedom per node (two in plane displacements)
- The resulting flat shell element has five degrees of freedom per node

Four-Noded Flat Shell Element -2-

- Stiffness matrix of the plate in the local coordinate system: $\underbrace{\tilde{\mathbf{k}}_{ben}}_{12 \times 12 \text{ matrix}}$
- Stiffness matrix of the plane stress element in the local coordinate system: $\underbrace{\tilde{\mathbf{k}}_{mem}}_{8 \times 8 \text{ matrix}}$
- Stiffness matrix of the flat shell element in the local coordinate system

$$\underbrace{\tilde{\mathbf{k}}}_{20 \times 20 \text{ matrix}} = \begin{bmatrix} \tilde{\mathbf{k}}_{mem} & \mathbf{0} \\ \mathbf{0} & \tilde{\mathbf{k}}_{ben} \end{bmatrix}$$

- Stiffness matrix of the flat shell element can be augmented to include the rotations $\tilde{\beta}_3$ (see figure on previous page)

$$\underbrace{\tilde{\mathbf{k}}^*}_{24 \times 24 \text{ matrix}} = \begin{bmatrix} \tilde{\mathbf{k}}_{mem} & \mathbf{0} & \mathbf{0} \\ \mathbf{0} & \tilde{\mathbf{k}}_{ben} & \mathbf{0} \\ \mathbf{0} & \mathbf{0} & \mathbf{0} \end{bmatrix}$$

- Stiffness components corresponding to $\tilde{\beta}_3$ are zero because neither the plate nor the plane stress element has corresponding stiffness components

Four-Noded Flat Shell Element -3-

- Transformation of the element stiffness matrix from the local to the global coordinate system

- Discrete element equilibrium equation in the local coordinate system

$$\tilde{k}^* \tilde{u} = \tilde{f}$$

- Nodal displacements and rotations of element \tilde{u}
- Element force vector \tilde{f}

- Transformation of vectors from the local to the global coordinate system

$$u = c\tilde{u} \quad f = c\tilde{f}$$

- Rotation matrix (or also known as the direction cosine matrix) c
- Note that for all rotation matrices $c^{-1} = c^T$

- Transformation of element stiffness matrix from the local to global coordinate system

$$\tilde{k}^* \tilde{u} = \tilde{f} \quad \Rightarrow \quad \tilde{k}^* c^T u = c^T f \quad \Rightarrow \quad \underbrace{c \tilde{k}^* c^T}_{k^*} u = f$$

- Discrete element equilibrium equation in the global coordinate system

$$k^* u = f$$

Four-Noded Flat Shell Element -4-

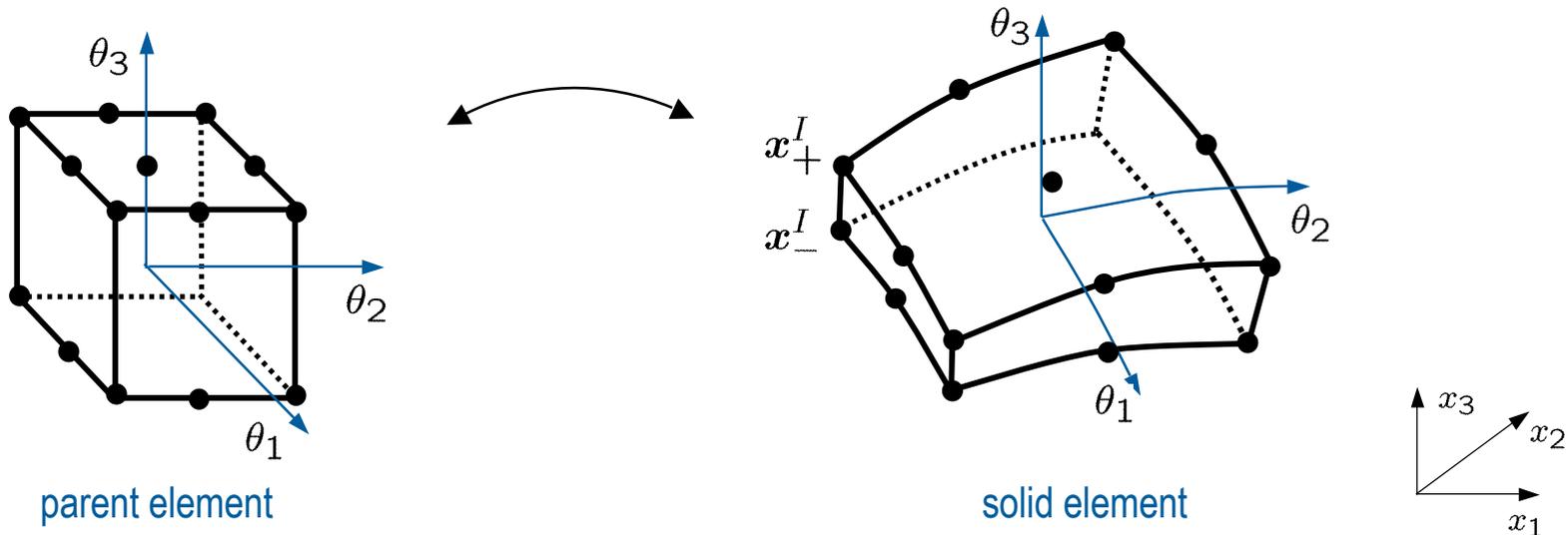
- The global stiffness matrix for the shell structure is constructed by transforming each element matrix into the global coordinate system prior to assembly
- The global force vector of the shell structure is constructed by transforming each element force vector into the global coordinate system prior to assembly
- Remember that there was no stiffness associated with the local rotation degrees of freedom $\tilde{\beta}_3$. Therefore, the global stiffness matrix will be rank deficient if all elements are coplanar.
 - It is possible to add some small stiffness for element stiffness components corresponding to $\tilde{\beta}_3$ in order to make global stiffness matrix invertible

$$\underbrace{\tilde{\mathbf{k}}^*}_{24 \times 24 \text{ matrix}} = \begin{bmatrix} \tilde{\mathbf{k}}_{mem} & 0 & 0 \\ 0 & \tilde{\mathbf{k}}_{ben} & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

Add small stiffness in order to make stiffness matrix invertible

Degenerated Shell Elements -1-

- First a three-dimensional solid element and the corresponding parent element are considered (isoparametric mapping)



- In the following it is assumed that the solid element has on its top and bottom surfaces nine nodes so that the total number of nodes is eighteen
 - The derivations can easily be generalised to arbitrary number of nodes
 - Coordinates of the nodes on the top surface are x_+^I for $I = 1, \dots, 9$
 - Coordinates of the nodes on the bottom surface are x_-^I for $I = 1, \dots, 9$

Degenerated Shell Elements -2-

- There are nine isoparametric shape functions for interpolating the top and bottom surfaces

$$N^I(\theta^1, \theta^2) \text{ for } I = 1, \dots, 9$$

- with the natural coordinates $\theta^1, \theta^2 \in (-1, +1)$
- Note that these shape functions are identical to the ones for two dimensional elasticity

- The geometry of the solid element can be interpolate with

$$\mathbf{x}(\theta_1, \theta_2, \theta_3) = \sum_{I=1}^9 N^I(\theta_1, \theta_2) \frac{1 + \theta_3}{2} \mathbf{x}_+^I + \sum_{I=1}^9 N^I(\theta_1, \theta_2) \frac{1 - \theta_3}{2} \mathbf{x}_-^I$$

with $-1 \leq \theta_3 \leq 1$

- Definitions

- Shell mid-surface node

$$\bar{\mathbf{x}}^I = \frac{1}{2}(\mathbf{x}_+^I + \mathbf{x}_-^I)$$

- Shell director (or fibre) at node I

$$\mathbf{d}^I = \frac{\mathbf{x}_+^I - \mathbf{x}_-^I}{\|\mathbf{x}_+^I - \mathbf{x}_-^I\|}$$

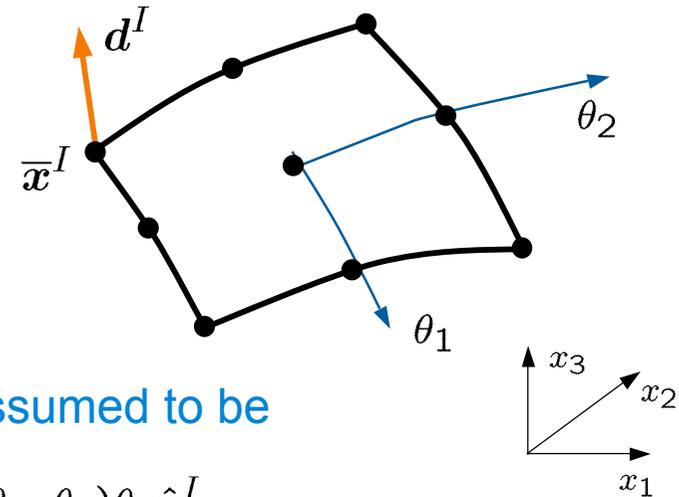
- The shell director is a unit vector and is approximately orthogonal to the mid-surface

Degenerated Shell Elements -3-

- Using the previous definitions the solid element geometry can be interpolated with

$$x(\theta_1, \theta_2, \theta_3) = \sum_{I=1}^9 N^I(\theta_1, \theta_2) \bar{x}^I + \frac{t}{2} \sum_{I=1}^9 N^I(\theta_1, \theta_2) \theta_3 d^I$$

- with the solid element thickness t



- The displacements of the solid element are assumed to be

$$u(\theta_1, \theta_2, \theta_3) = \sum_{I=1}^9 N^I(\theta_1, \theta_2) \bar{u}^I + \frac{t}{2} \sum_{I=1}^9 N^I(\theta_1, \theta_2) \theta_3 \hat{u}^I$$

- The first component is the mid-surface displacement and the second component is the director displacement
 - Note that the deformed mid-surface nodal coordinates can be computed with $\bar{x}^I + \bar{u}^I$ and the deformed nodal director with $d^I + \hat{u}^I$
- The director displacement \hat{u}^I has to be constructed so that the director d^I can rotate but not stretch
 - This was one of the of the Reissner-Mindlin theory assumptions

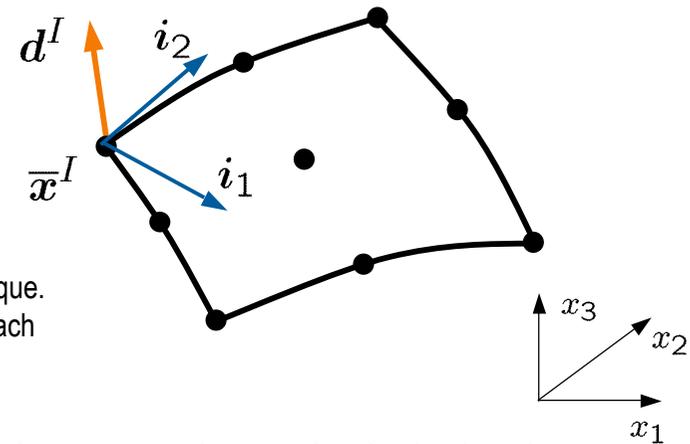
Degenerated Shell Elements -4-

- The director displacements are expressed in terms of rotations at the nodes

- To accomplish this a local orthonormal coordinate system $i_1 - i_2$ is constructed at each node

- $i_1 \cdot i_2 = 0$
 - $i_1 \cdot i_1 = 1$ and $i_2 \cdot i_2 = 1$
 - $i_1 \cdot d^I = 0$ and $i_2 \cdot d^I = 0$

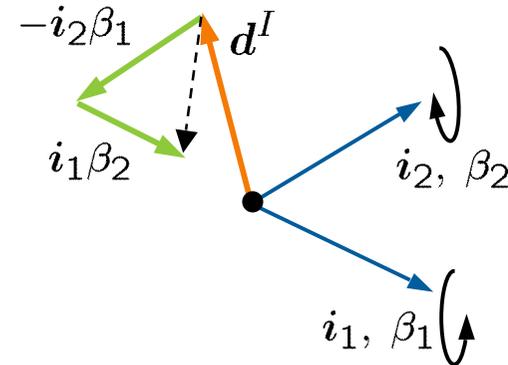
- The definition of the orthonormal coordinate system is not unique. In a finite element implementation it is necessary to store at each node the established coordinate base vectors $i_1 - i_2$.



- The relationship between the director displacements and the two rotation angles in the local coordinate system is

$$\hat{u}^I = -i_2 \beta_1^I + i_1 \beta_2^I$$

- Rotations β_1 and β_2 are defined as positive with the right-hand rule
- It is assumed that the rotation angles are small
- $\hat{u}^I \cdot d^I = 0$ so that the director length does not change



Degenerated Shell Elements -5-

- Displacement of the shell element in dependence of the mid-surface displacements and director rotations

$$\mathbf{u}(\theta_1, \theta_2, \theta_3) = \sum_{I=1}^9 N^I \bar{\mathbf{u}}^I + \frac{t}{2} \sum_{I=1}^9 N^I \theta_3 (-i_2 \beta_1^I + i_1 \beta_2^I)$$

- The element has nine nodes
 - There are five unknowns per node (three mid-surface displacements and two director rotations)
 - This assumption about the possible displacements is equivalent to the Reissner-Mindlin assumption
- Introducing the displacements into the strain equation of three-dimensional elasticity leads to the strains of the shell element

$$\epsilon_{ij} = \frac{1}{2} \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right)$$

- In computing the displacement derivatives the chain rule needs to be used

$$\frac{\partial u_i}{\partial x_j} = \frac{\partial u_i}{\partial \theta_k} \frac{\partial \theta_k}{\partial x_j} \quad \text{with} \quad \frac{\partial \theta_k}{\partial x_j} = \left(\frac{\partial x_j}{\partial \theta_k} \right)^{-1}$$

Jacobian

Degenerated Shell Elements -6-

- The Jacobian is computed from the geometry interpolation

$$\mathbf{x}(\theta_1, \theta_2, \theta_3) = \sum_{I=1}^9 N^I(\theta_1, \theta_2) \bar{\mathbf{x}}^I + \frac{t}{2} \sum_{I=1}^9 N^I(\theta_1, \theta_2) \theta_3 \mathbf{d}^I$$

$$\frac{\partial \mathbf{x}}{\partial \theta_\alpha} = \sum_{I=1}^9 \frac{\partial N^I}{\partial \theta_\alpha} \bar{\mathbf{x}}^I + \frac{t}{2} \sum_{I=1}^9 \frac{\partial N^I}{\partial \theta_\alpha} \theta_3 \mathbf{d}^I \quad \text{for } \alpha = 1, 2$$

$$\frac{\partial \mathbf{x}}{\partial \theta_3} = \frac{t}{2} \sum_{I=1}^9 N^I \mathbf{d}^I$$

- The shell strains introduced into the internal virtual work of three-dimensional elasticity give the internal virtual work of the shell
- For shear locking similar techniques such as developed for the Reissner-Mindlin plate need to be considered